



MDA BOOSTER COST ANALYSIS

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MDA Boosters – Cost Analysis Seminar

- ***MDA Booster Missions & Requirements***
 - *Interceptors*
 - *Target Systems*

- ***Common Booster Elements & Components***
 - *SRMs & TVC Nozzles*
 - *Interstage Structures*
 - *Electrical Power*
 - *Guidance & Control Electronics*
 - *G&C Software*
 - *Safety Systems*
 - *Payload Systems*

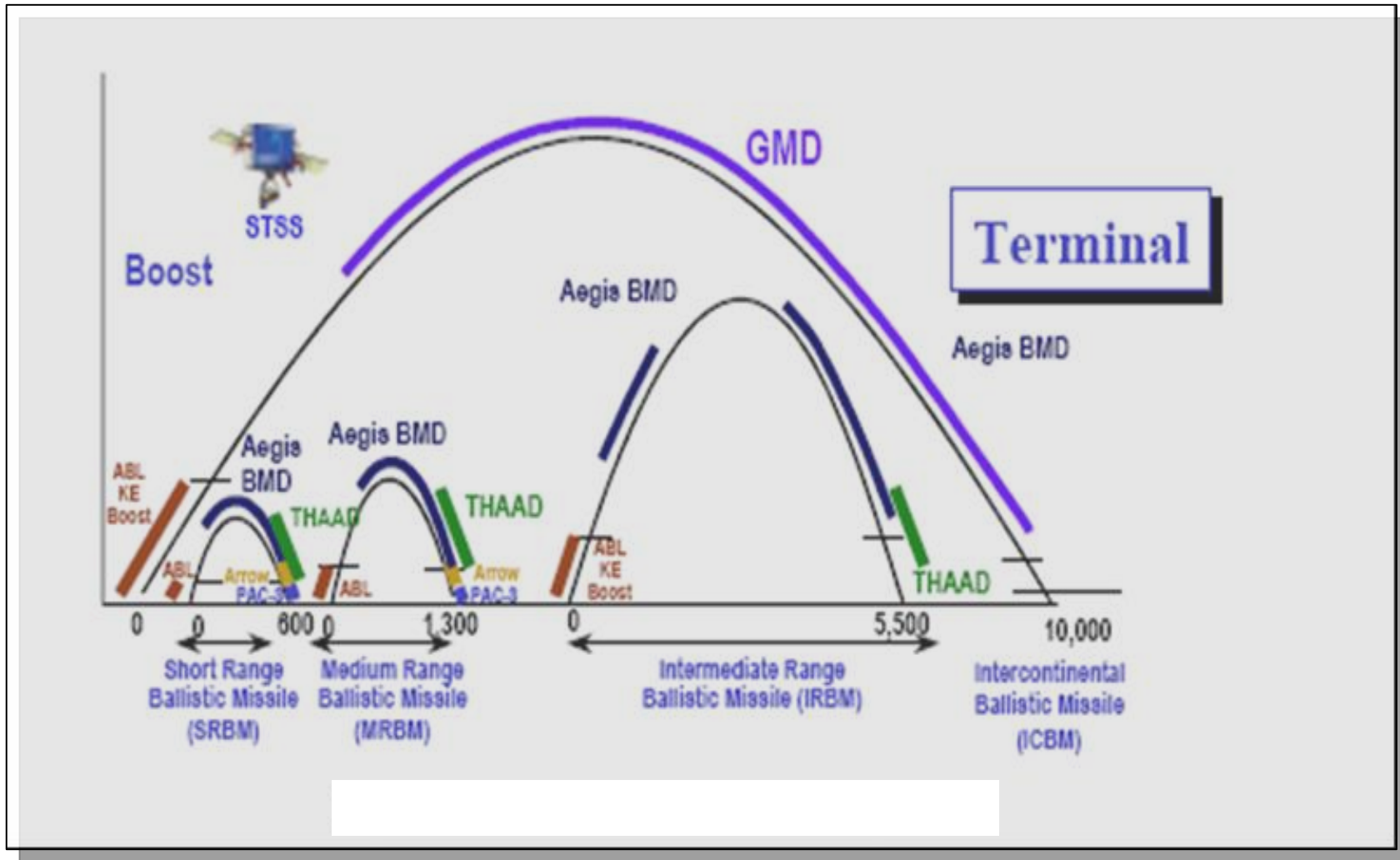
- ***Basing, Support Systems & Below-the-Line Costs***

- ***Launch Ops and Range Operation Costs***

- ***Conclusions***



Missile Defense Agency Interceptor Systems





Missile Defense Agency Interceptor Systems



MDA Boosters - Interceptors



Kinetic Energy Interceptor



Standard Missile-3



Ground-Based Interceptor Midcourse Defense

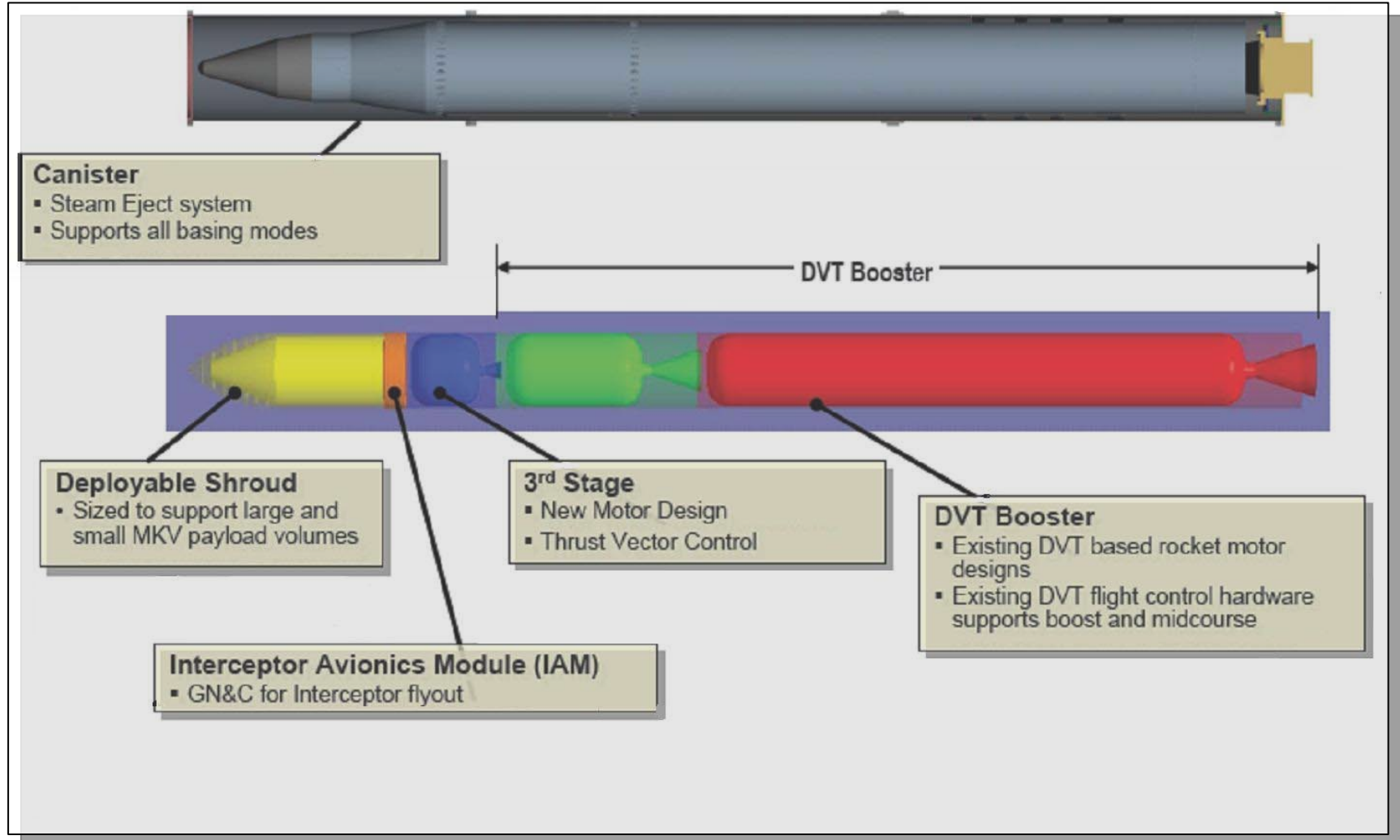


Terminal High Altitude Area Defense (THAAD)



Patriot Advanced Capability-3 (PAC3)

Missile Defense Agency Interceptor Systems - KEI



Missile Defense Agency Interceptor Systems - KEI



INTERCEPTOR:	KINETIC ENERGY INTERCEPTOR
STATUS:	UNDER DEVELOPMENT
PRIME CONTRACTOR	NORTHROP GRUMMAN
SEEKER	MULTIPLE INFRARED KILL VEHICLES
PROPULSION TYPE	SRM
GROSS LIFT OFF WGT (LBS)	26500+
STAGES	3 STAGES + KILL VEHICLES
LENGTH (FT)	38.8
MAX SPAN - WINGS/FINS (FT)	3.33
BODY DIAMETER (FT)	3.33
FORCE SIZE	TBD
IOC	POST 2013
MISSION	BOOST PHASE ANTIMISSILE
EFFECTIVE RANGE (NM)	750+

Missile Defense Agency Interceptor Systems – SM3



Missile Defense Agency Interceptor Systems – SM3



INTERCEPTOR:	STANDARD MISSILE 3 (RIM-161A)
STATUS:	OPERATIONAL
PRIME CONTRACTOR	RAYTHEON
SEEKER	INFRARED KILL VEHICLE
PROPULSION TYPE	SRM
GROSS LIFT OFF WGT (LBS)	3310
STAGES	3 STGS, 1 KILL VEHICLE
LENGTH (FT)	21.6
MAX SPAN - WINGS/FINS (FT)	3.5
BODY DIAMETER (FT)	1.1
FORCE SIZE	1200
IOC	2002
MISSION	AREA DEFENSE - ANTIMISSILE
EFFECTIVE RANGE (NM)	270+

Missile Defense Agency Interceptor Systems - GBI

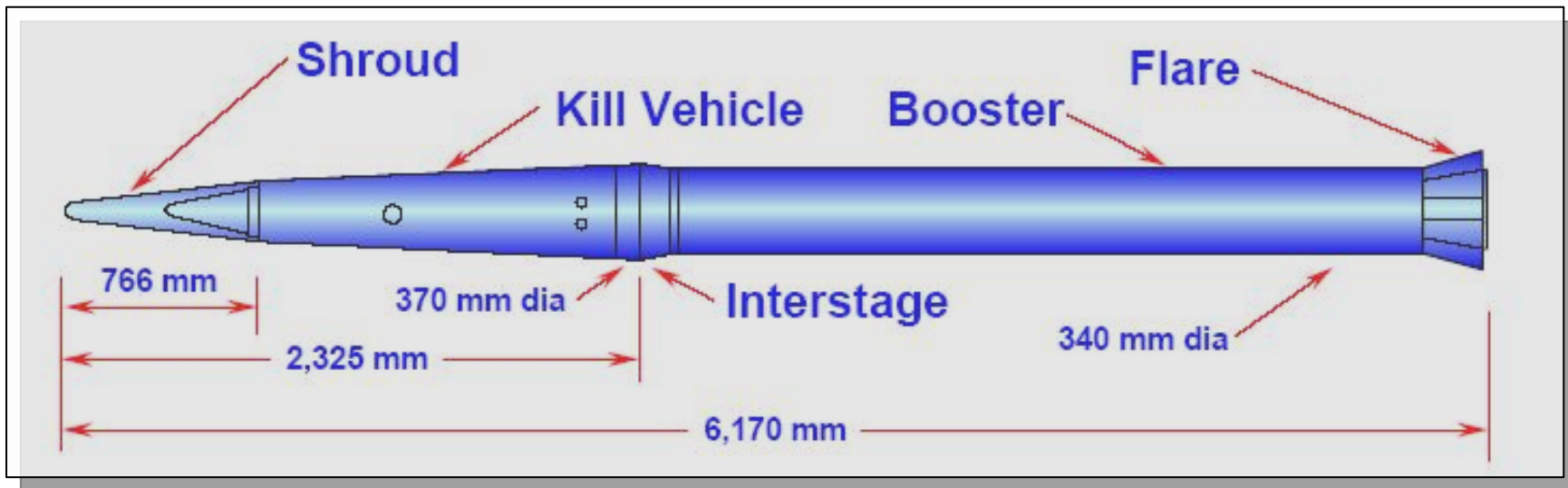


Missile Defense Agency Interceptor Systems - GBI



INTERCEPTOR:	GROUND BASED INTERCEPTOR
STATUS:	OPERATIONAL
PRIME CONTRACTOR	BOEING
SEEKER	INFRARED KILL VEHICLE
PROPULSION TYPE	SRM
GROSS LIFT OFF WGT (LBS)	28000+
STAGES	3 STAGES + KILL VEHICLE
LENGTH (FT)	55
MAX SPAN - WINGS/FINS (FT)	4.17
BODY DIAMETER (FT)	4.17
FORCE SIZE	50+
IOC	2003
MISSION	MIDCOURSE ANTIMISSILE
EFFECTIVE RANGE (NM)	3300+

Missile Defense Agency Interceptor Systems - THAAD

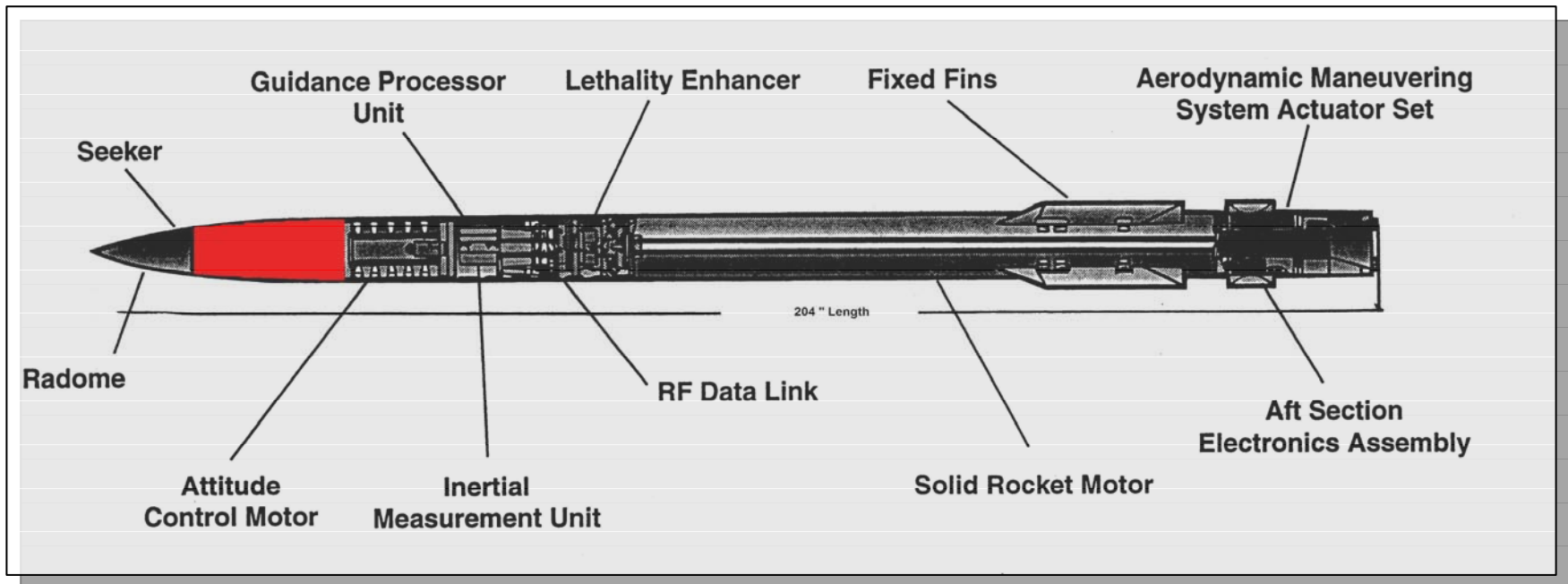


Missile Defense Agency Interceptor Systems - THAAD



INTERCEPTOR:	THAAD
STATUS:	UNDER DEVELOPMENT
PRIME CONTRACTOR	LOCKHEED MARTIN
SEEKER	ACTIVE/SEMI-ACTIVE RADAR
PROPULSION TYPE	SRM
GROSS LIFT OFF WGT (LBS)	2000
STAGES	1 BOOSTER, 1 HTK DACS
LENGTH (FT)	20.4
MAX SPAN - WINGS/FINS (FT)	-----
BODY DIAMETER (FT)	1.1
FORCE SIZE	48
IOC	2009
MISSION	TERMINAL ANTIMISSILE
EFFECTIVE RANGE (NM)	125+

Missile Defense Agency Interceptor Systems – PAC3



Missile Defense Agency Interceptor Systems – PAC3

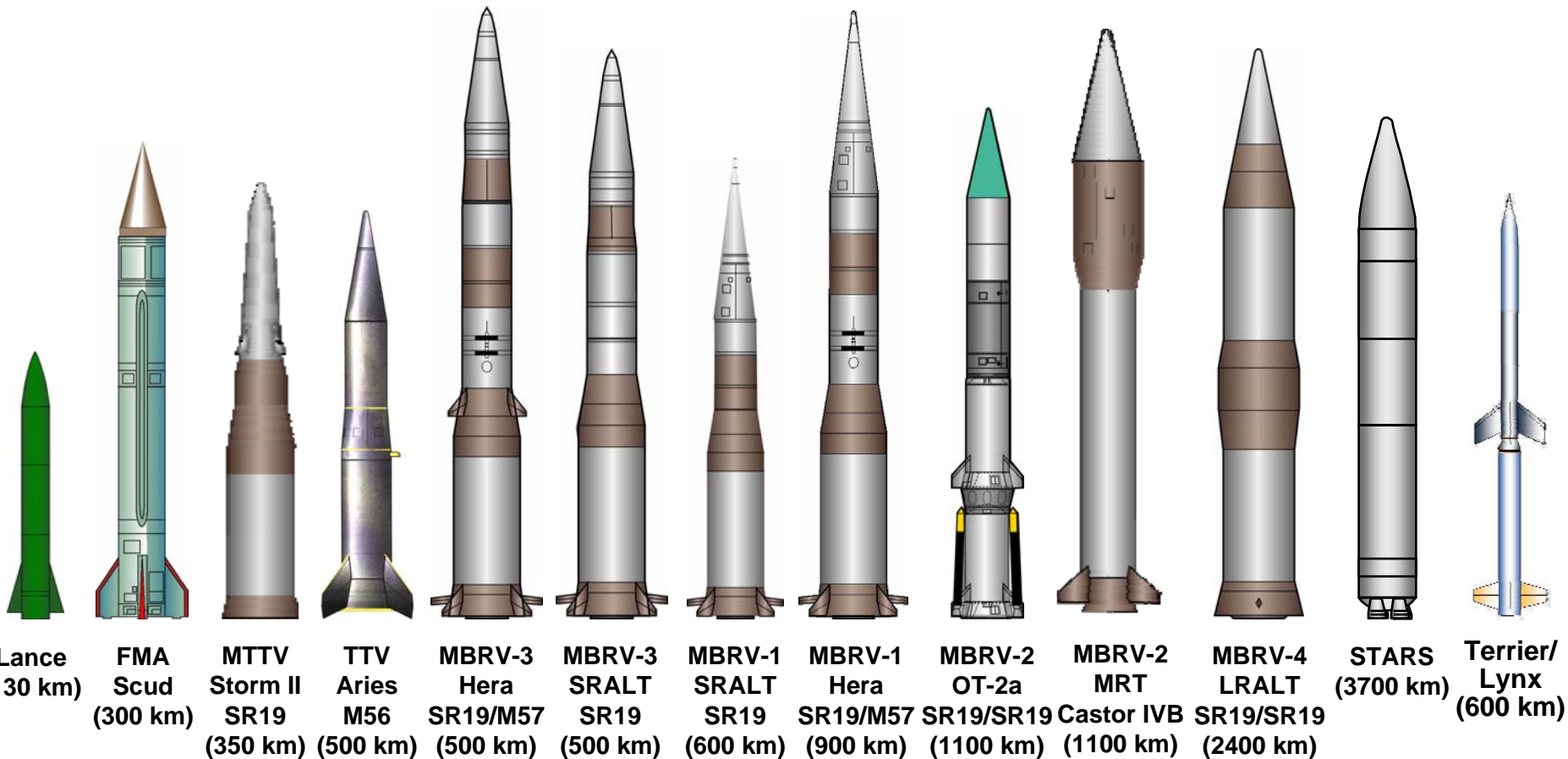


INTERCEPTOR:	PAC-3
STATUS:	OPERATIONAL
PRIME CONTRACTOR	LOCKHEED MARTIN
SEEKER	ACTIVE/SEMI-ACTIVE RADAR
PROPULSION TYPE	SRM
GROSS LIFT OFF WGT (LBS)	700
STAGES	1 BOOSTER + MANEUVER KILL VEHICLE
LENGTH (FT)	17
MAX SPAN - WINGS/FINS (FT)	1.67
BODY DIAMETER (FT)	1
FORCE SIZE	1200+
IOC	1995
MISSION	TERMINAL ANTIMISSILE
EFFECTIVE RANGE (NM)	25+



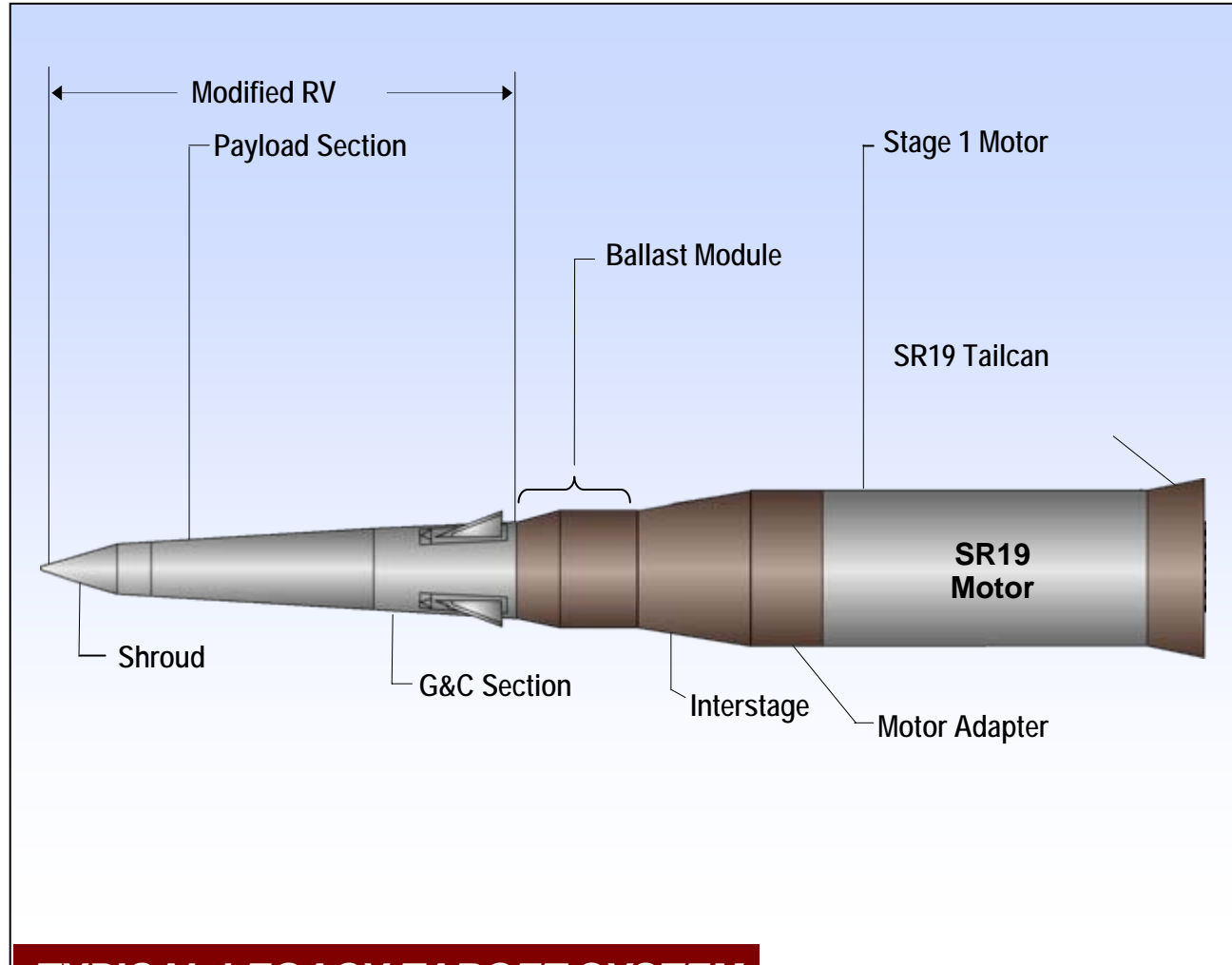
Missile Defense Agency Target Systems

Family of Legacy Targets (52 Inch Diameter and Smaller)





Missile Defense Agency Target Systems

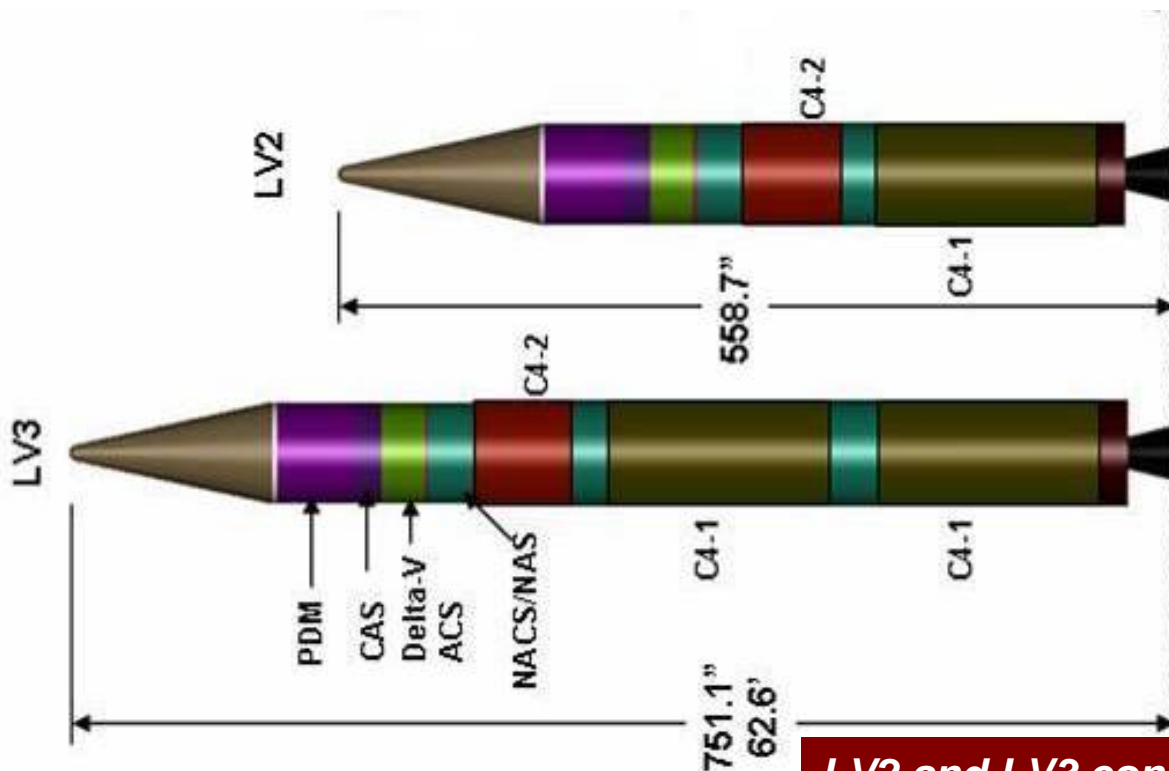


TYPICAL LEGACY TARGET SYSTEM



Missile Defense Agency Target Systems

Family of 72" Targets (Under Development):



LV2 and LV3 concepts use retired C4 SRMS and modular interstages, payload instruments, and countermeasure packages.



Missile Parametric Cost Model

- **DEVELOPED BY GREG BELL FOR USE ON SPECIAL MISSILE, MUNITION, AND MDA CONCEPT STUDIES & PROGRAMS.**
- **THE MODEL HAS UNDERGONE CONTINUOUS REFINEMENT SINCE 1998, AND HAS GROWN INTO A FAMILY OF VERSIONS COVERING VARIOUS TACTICAL, STRATEGIC, AND MDA MISSIONS AND TECHNOLOGIES.**
- **THE MODEL IS A PARAMETRIC MODEL WITH CERS, FACTORS, AND DEFAULT VALUES THAT ARE USED TO ESTIMATE ALL LCC ELEMENTS, INCLUDING DEVELOPMENT, PRODUCTION, SUPPORT, AND O&S COSTS.**
- **PARAMETRIC METHODS (Examples):**

$$\text{COST1} = f(\text{WEIGHT})$$

$$\text{COST2} = f(\text{KSLOC})$$

$$\text{COST3} = f(\text{EVENTS or OBJECTS})$$

$$\text{COST4} = f(\text{OTHER COST})$$

$$\text{COST5} = f(\text{THRUST}/F_n)$$

$$\text{COST6} = f(\text{SHP})$$

$$\text{COST7} = f(\text{POWER} * \text{APERTURE})$$

and Many Others



Cost Model Description

- ***Populated with historical data from U.S. industry programs***
- ***CERs forecast labor hours and material dollars, based on weight, K-SLOC, test events, Engine power output, and other independent drivers.***
- ***Global Inputs: wrap rates, fly-to-buy ratios, equipment and raw material costs/lb, and adjustment factors for composites and specialty metals (aluminum is a 1.0).***
- ***Direct inputs: Weights, Test Agenda and Hours of Test Flight, software KSLOC, engine characteristics, programmatic planning and special factors for complexity.***
- ***Model generates hours for design, test, tooling, factory, SEPM, QA, ILS, other support efforts, material and supplier costs and integrates them into a parametric cost estimate.***
- ***Over 400 CERs are available in MPCM.***

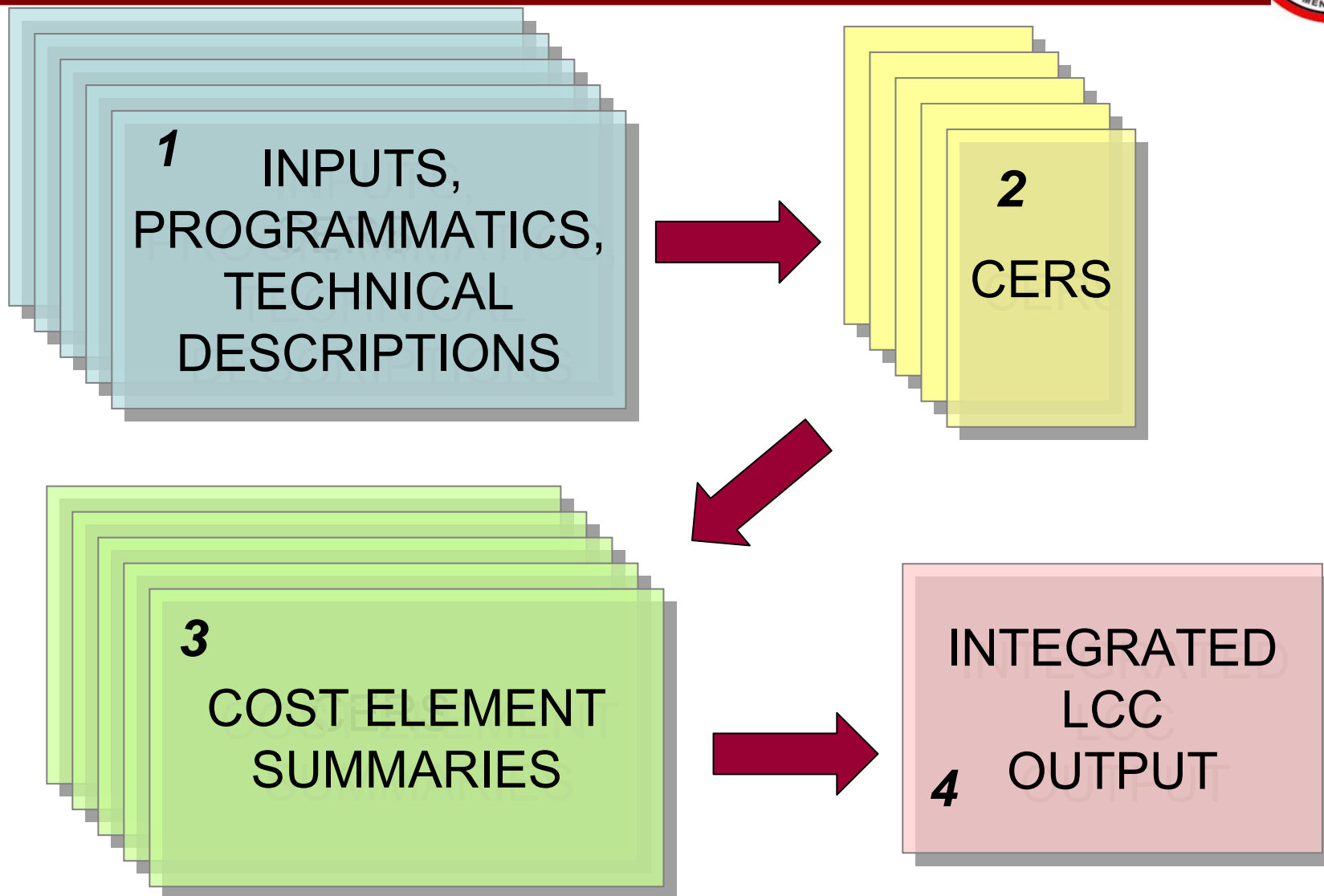


Some Model Sources

- *"Cost Models and Support Data for Solid Propellant Rocket Motors (U)," Tecolote Research, Inc., TM-127, 1980.*
- *"Electro-optical, Missile, Radar, and Avionics Systems Cost Research, Volume 3, Missile Production Cost Model," MCR, Inc., 1988.*
- *"Missile Cost Handbook," Nicholas, T., et. al., Data Search Associates, Fountain Valley, 1976-1998.*
- *"Naval Weapons Center (NWC) Missile Cost Model," A. Vokolek, NWC 1983.*
- *"Naval Weapons Center (NWC) Modular Missile Cost Model," Tecolote Research Inc., CR-0172, 1987.*
- *"R&D Cost Model for Tactical Missile Solid Propulsion Systems," Tecolote Research, Inc., TM-158, 1982.*
- *"Solid Rocket Motor Cost/Reliability Predicting Methodology," Booz Allen, NWC TP 5634, NWC 1974.*
- *"Tactical Missile Development Cost," SAIC, 1987.*
- *"Tactical Missile RDT&E Cost Model," Tecolote Research, TM119, 1979.*



Parametric Cost Estimation Process





CER Libraries For Each Cost Element

MSL ENGINEERING CERS 12-Jul-03	"A" COEFF	"B" EXP	CER FORM	OBSER- VATIONS	R^2	SEE	RANGE OF INDE- PENDENT VARIABLE	SOURCE
DESIGN ENGINEERING (HDW)								
STRUCTURES DESIGN								
Wing Grp	344.0	1.00	DES MHRS = A*(WGT^B)	7		BASED ON MHRS/LB	1 LBS - 1000 LBS	INDUSTRY DATA BASE
Tail Grp	498.0	1.00	DES MHRS = A*(WGT^B)	7		BASED ON MHRS/LB	1 LBS - 1000 LBS	INDUSTRY DATA BASE
Fuselage Grp	612.3	1.00	DES MHRS = A*(WGT^B)	7		BASED ON MHRS/LB	1 LBS - 1000 LBS	INDUSTRY DATA BASE
Inlet Grp	78.5	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LBS - 100 LBS	
Intertank Structure	14.1	1.00	DES MHRS = A*(WGT^B)	3		BASED ON MHRS/LB	410 LBS - 13800 LBS	INDUSTRY DATA BASE
Fuel/Oxidizer Tank	12.9	1.00	DES MHRS = A*(WGT^B)	4		BASED ON MHRS/LB	700 LBS - 36000 LBS	INDUSTRY DATA BASE
Dome Grp (RF)	406.5	1.00	DES MHRS = A*(WGT^B)	5		BASED ON MHRS/LB	1 LB - 100 LBS	INDUSTRY DATA BASE
Dome Grp (EO)	3359.0	1.00	DES MHRS = A*(WGT^B)	4		BASED ON MHRS/LB	1 LBS - 100 LBS	INDUSTRY DATA BASE
RAM/RAS Structures	24.8	1.08	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LBS - 100 LBS	
Other Structures	0.0	1.00	DES MHRS = A*(WGT^B)					
PROPULSION GROUP DESIGN								
Liquid Rocket Engine	44.4	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 1000 LBS	
Solid Rocket Motor	44.4	1.00	DES MHRS = A*(WGT^B)	8		BASED ON MHRS/LB	10 LBS - 2000 LBS	INDUSTRY DATA BASE
Exhaust/Nozzle Grp	78.4	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Turbojet Engine	44.4	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 1000 LBS	
Turbofan Engine	44.4	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 1000 LBS	
Ramjet (External)	182.8	1.00	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 1000 LBS	
Ramjet (Internal)	182.8	1.00	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 1000 LBS	
Engine Start/Cntl	176.0	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Engine Accessories	176.0	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Fuel/Lubrication Grp	197.6	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Pressure Tank/Lines	74.5	1.00	DES MHRS = A*(WGT^B)	2		BASED ON MHRS/LB	650 LBS - 9500 LBS	INDUSTRY DATA BASE
Other Propulsion	0.0	1.00	DES MHRS = A*(WGT^B)					
FIXED EQUIP DESIGN								
Surface Cntls	1367.0	1.00	DES MHRS = A*(WGT^B)	7		BASED ON MHRS/LB	1 LB - 100 LBS	INDUSTRY DATA BASE
Aux Power Grp (SSG)	182.2	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Hyd & Pneu Grp	345.2	0.85	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Electrical Grp	1403.0	1.00	DES MHRS = A*(WGT^B)	8		BASED ON MHRS/LB	1 LB - 100 LBS	INDUSTRY DATA BASE
Environ Cntl Grp	262.2	1.00	DES MHRS = A*(WGT^B)			POSTULATED TREND	1 LB - 100 LBS	
Other Fxd Eq Grp	0.0	1.00	DES MHRS = A*(WGT^B)					

Example



WBS for Booster Cost Estimating

**Development
Cost**

**Production
Cost**

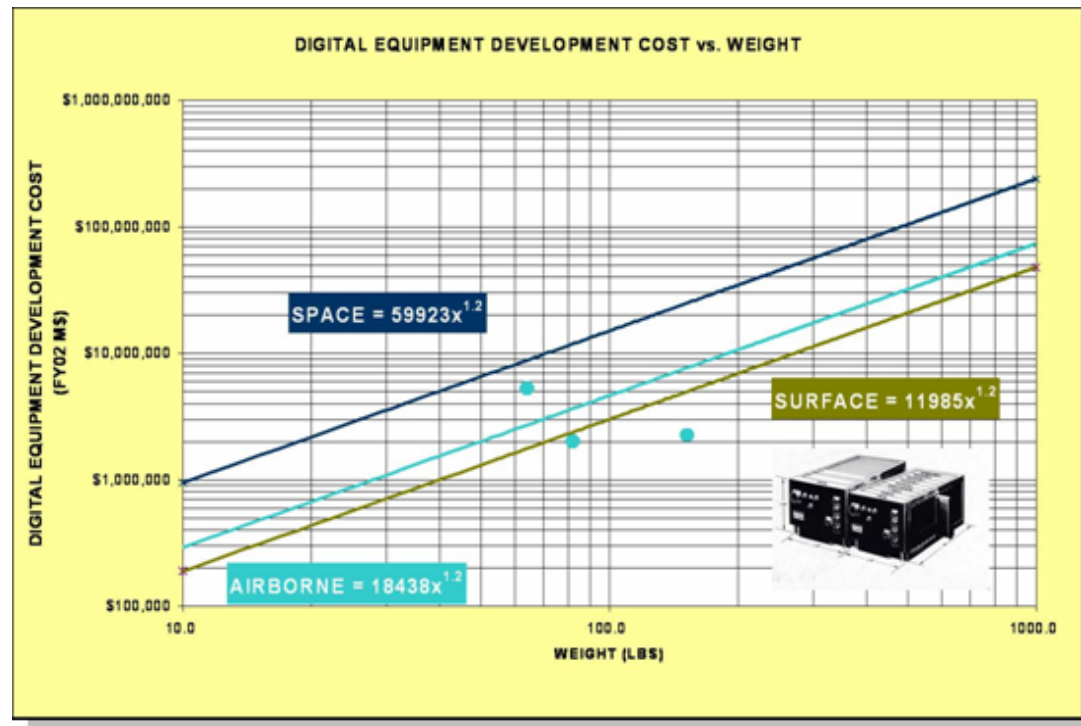
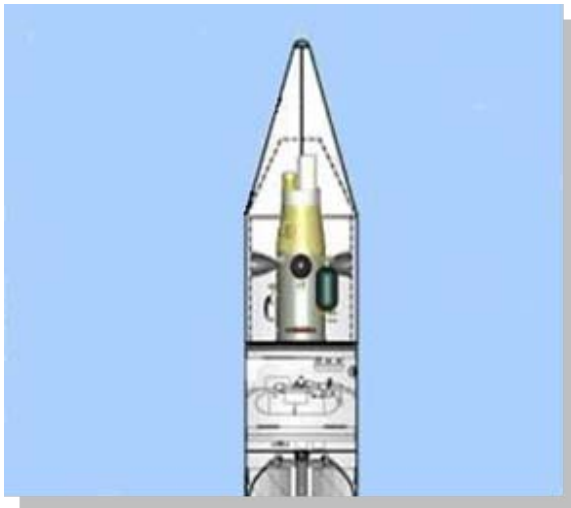
Prime Mission Product

- SRMs & TVC Nozzles*
- Interstage Structures*
- Electrical Power*
- Guidance & Cntl Electronics*
- G&C Software*
- Safety Systems*
- Payload Systems*

Below The Line Costs

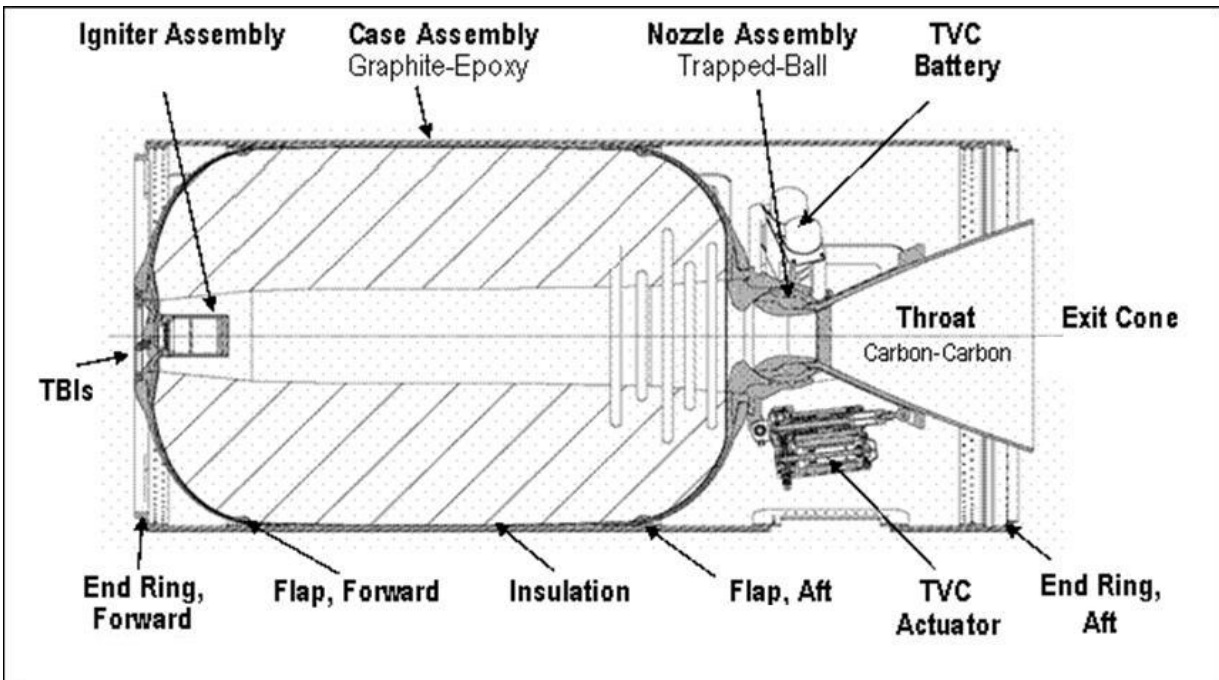
- Ground Support Equip (Pad)*
- Ground Support Equip (Transport)*
- Data*
- Training*
- Pad Spares*
- Site Activation*

Missile Technology & Cost Estimating Methods

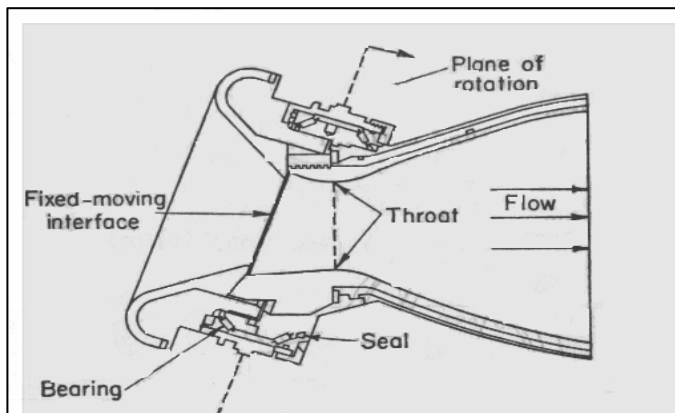




SRMs and TVC Nozzles

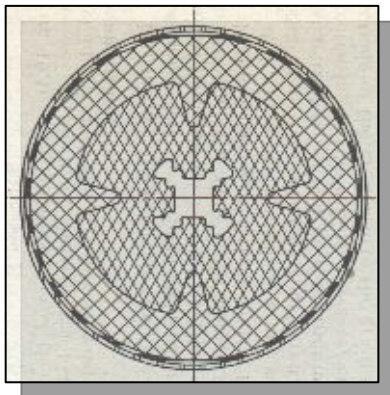
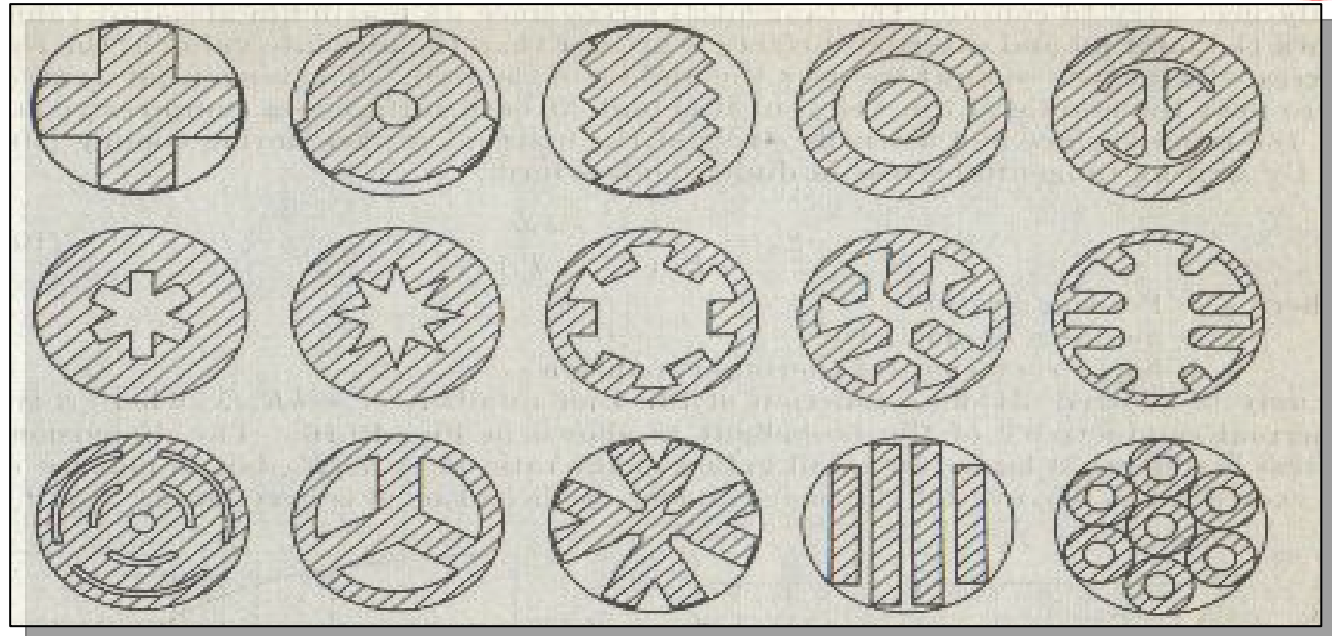


Solid Rocket Motors consist of a case, insulation, poured propellant (called a grain), igniter, and a Nozzle system. The exposed surface area of the Grain determines the burn rate and specific thrust.



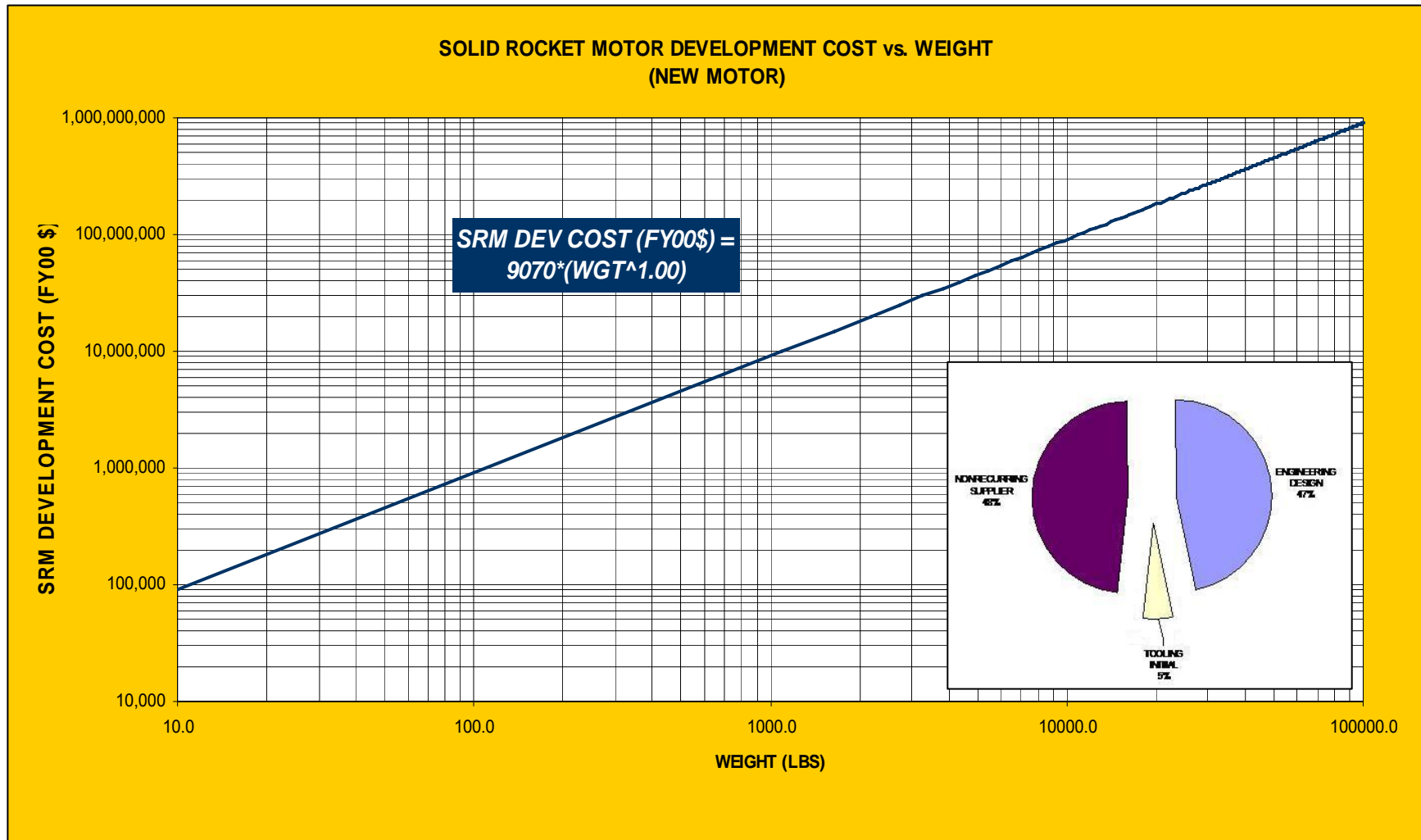


SRMs and TVC Nozzles – Propellant Grain

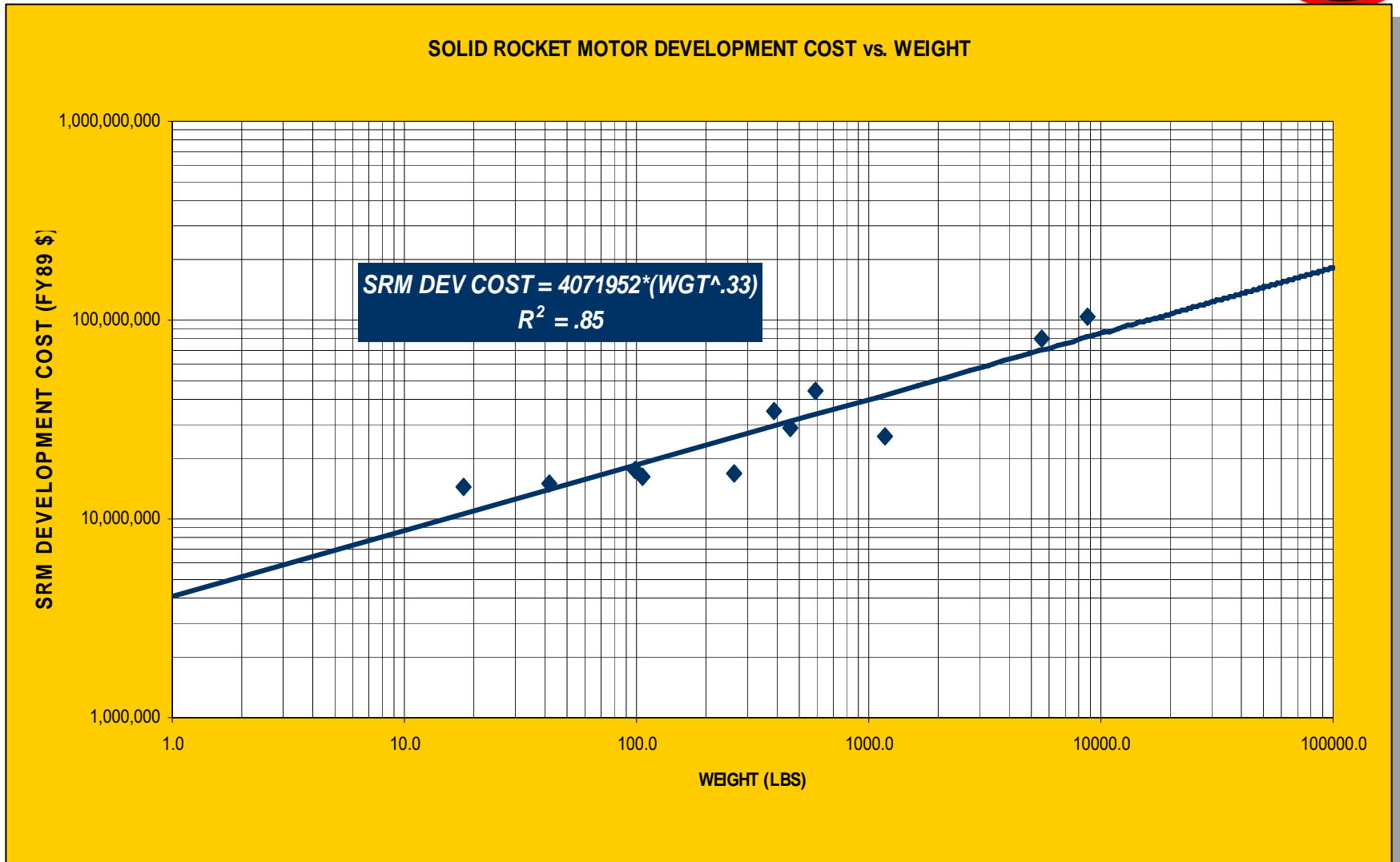


Propellant cross-section determines the burn rate and specific thrust. These figures illustrate some historical cross-sections and how a dual grain can be used to provide high boost thrust followed by low sustainer thrust.

SRMs and TVC Nozzles – Development Cost (New Motors)



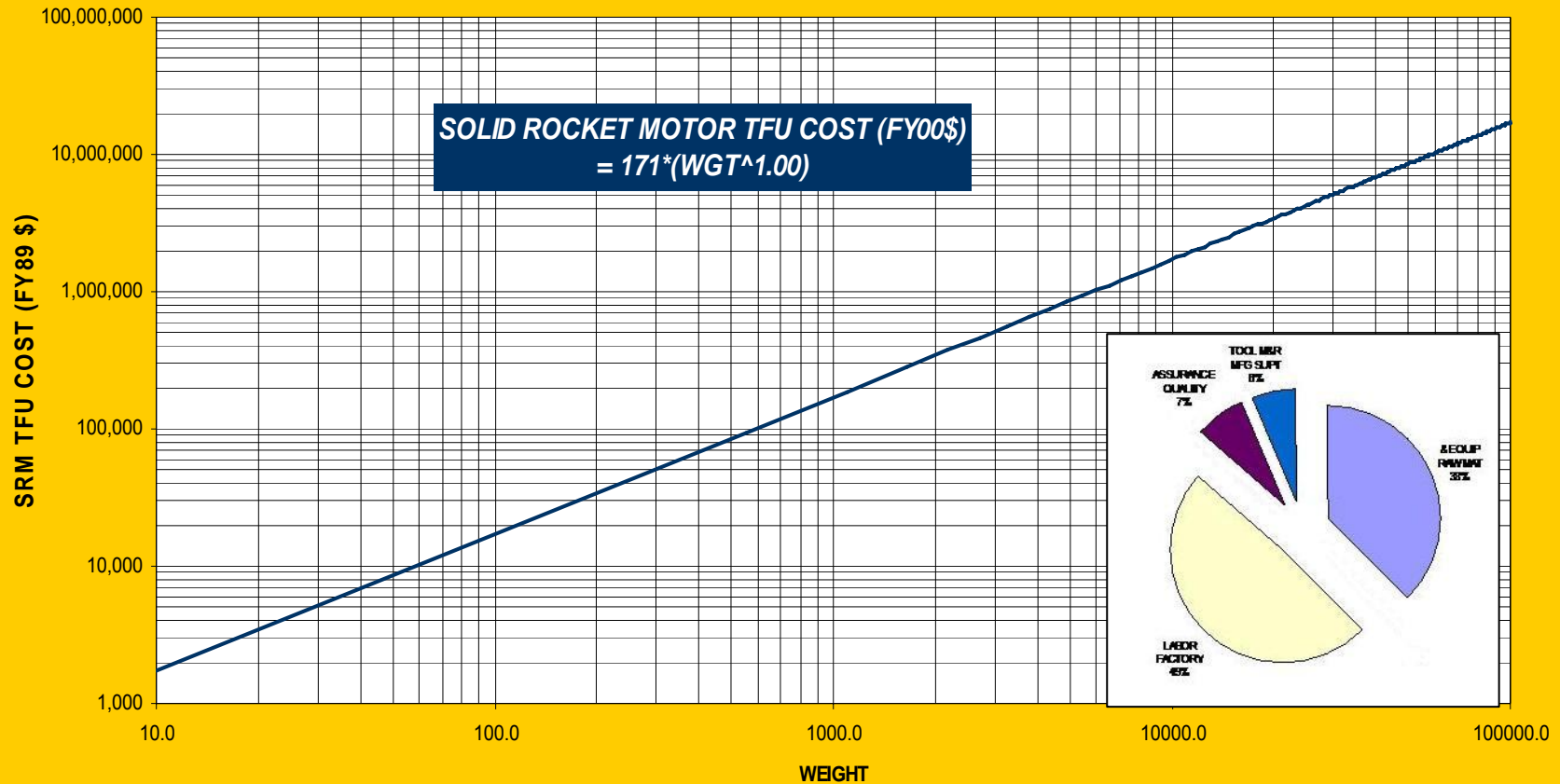
SRMs and TVC Nozzles – Development Cost (Basis of Estimate)



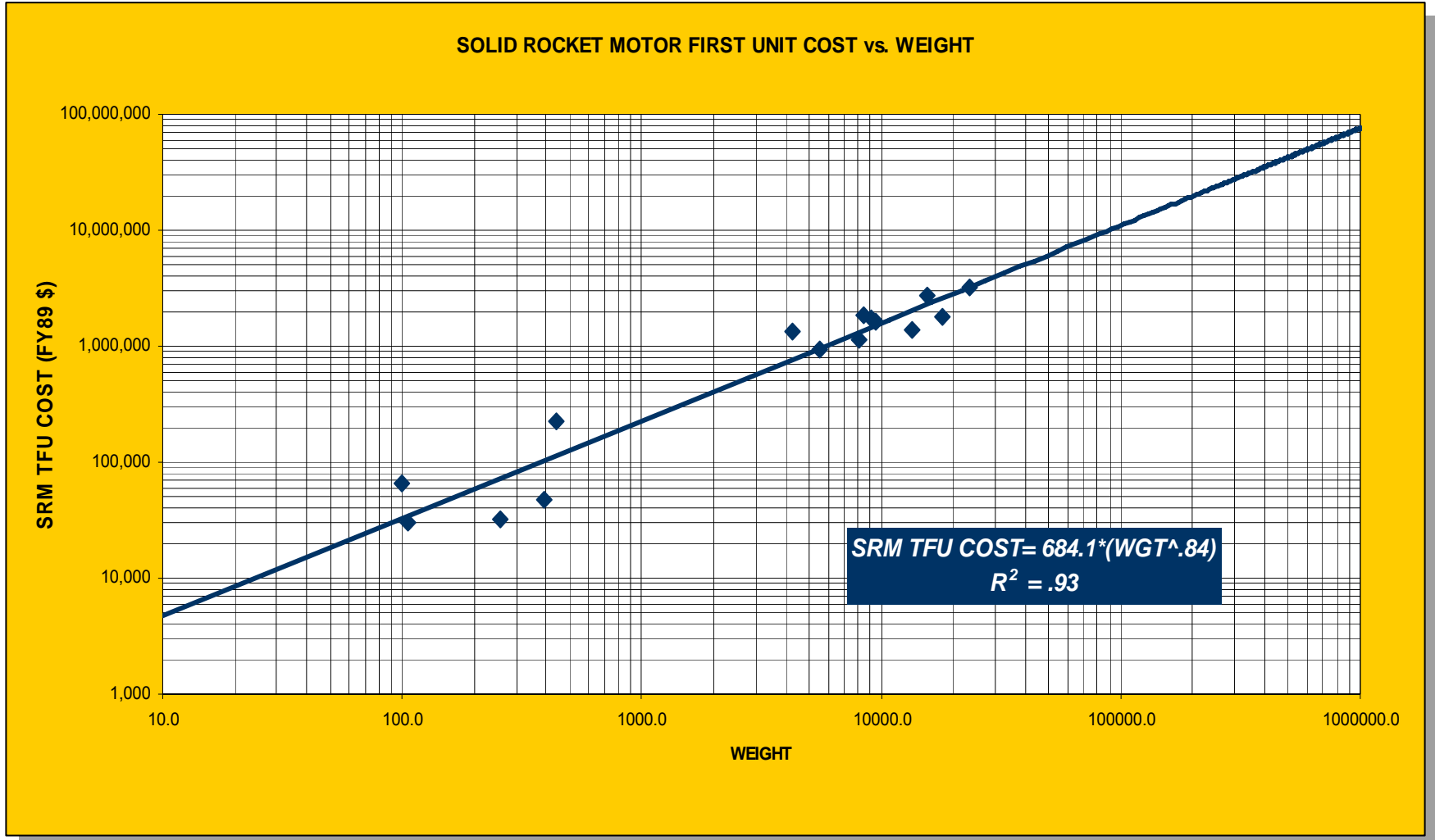
SRMs and TVC Nozzles (New Motors) TFU Production Cost



SOLID ROCKET MOTOR FIRST UNIT COST vs. WEIGHT

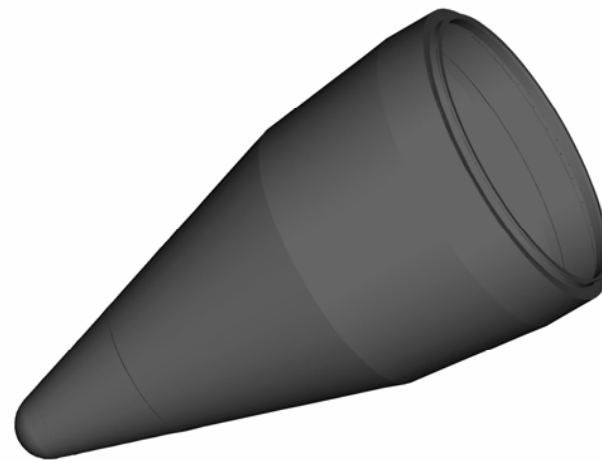
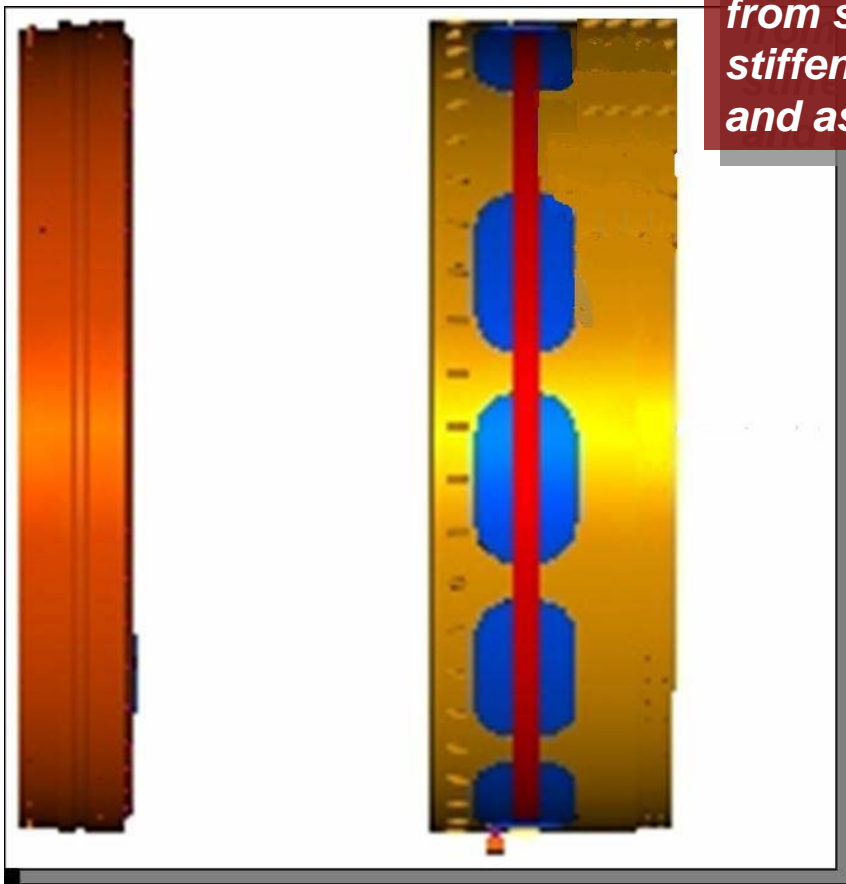


SRMs and TVC Nozzles (New Motors) TFU Production Cost (Basis of Estimate)



Interstage and Shroud Structures

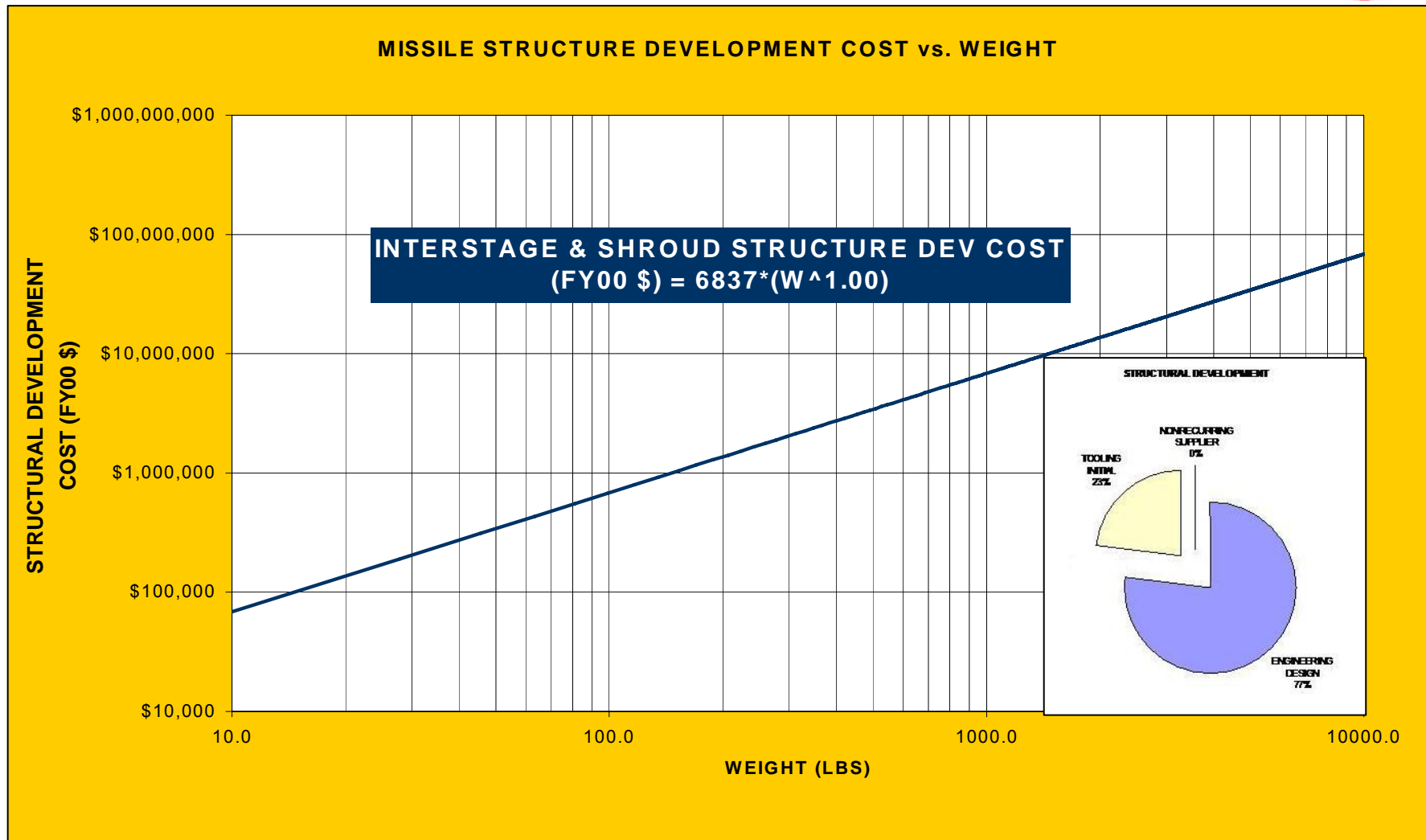
Typical interstage structures are fabricated from steel. They include skins, frames, stiffeners and doors to allow installation and assembly of stages and components.



Shrouds are fabricated from titanium, high temp steel, and may have ablative coatings.



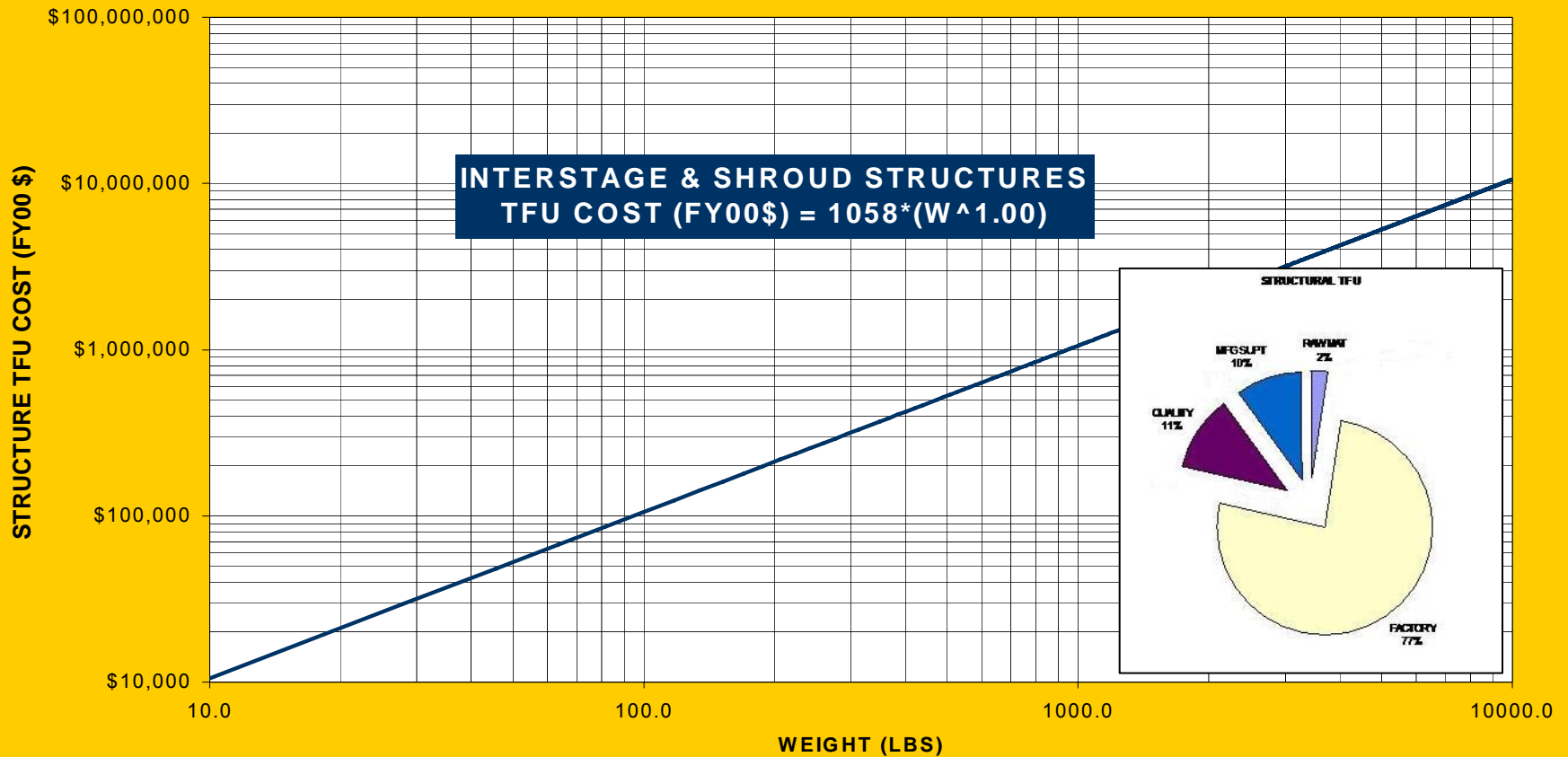
Structures – Development Cost



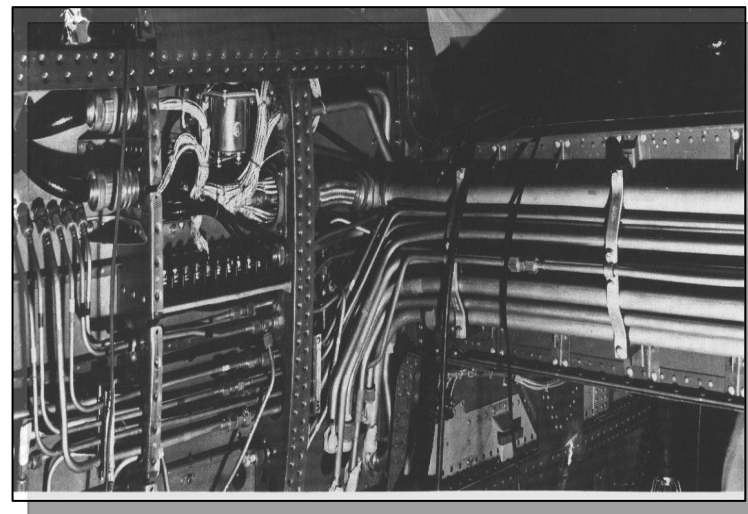


Structures – TFU Production Cost

MISSILE STRUCTURE
FIRST UNIT PRODUCTION COST vs. WEIGHT



Electrical Power – Batteries and PC&D

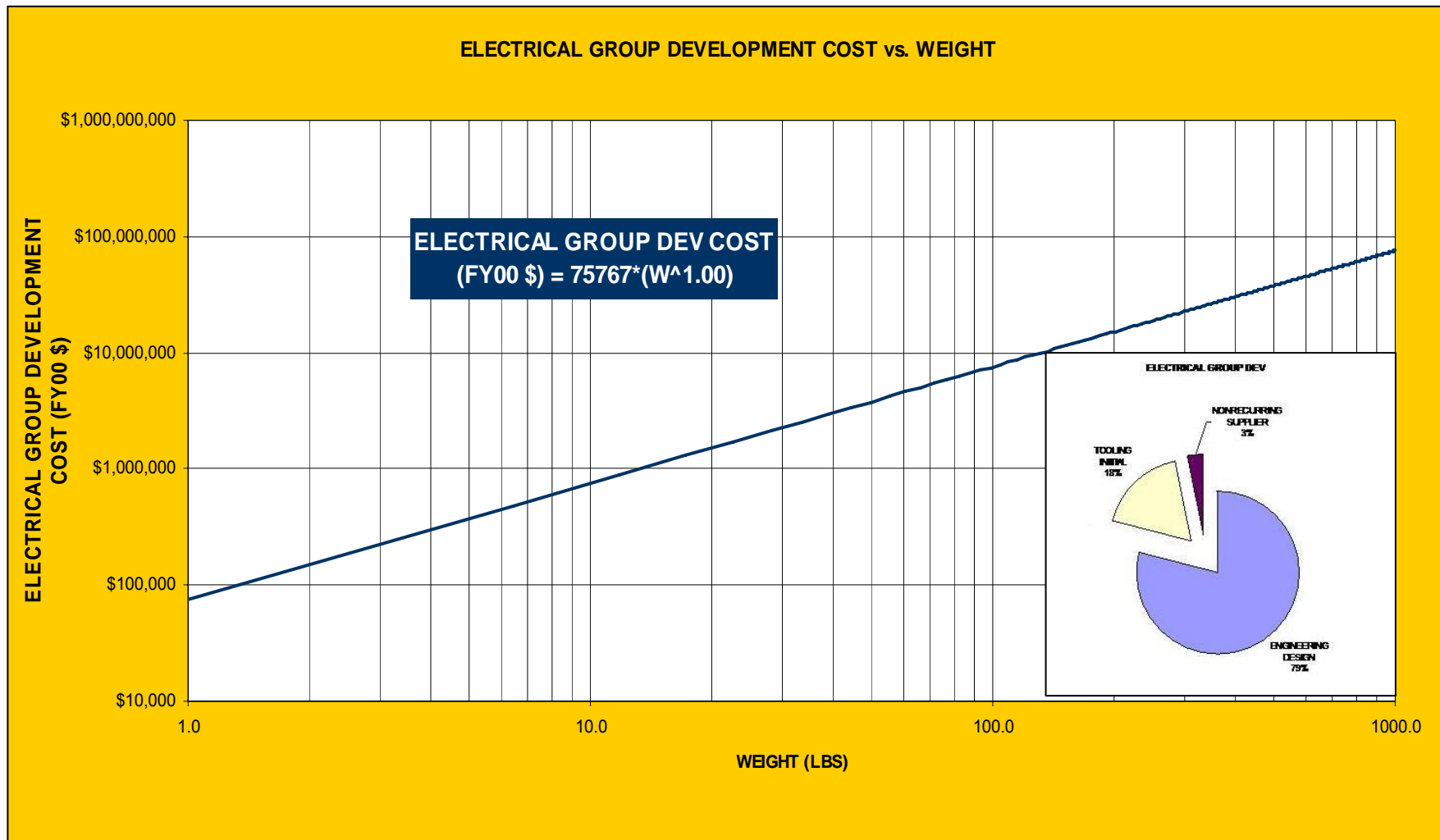


Electrical power for missiles generally comes from banks of thermal batteries. For the largest ICBMs and ELVs, chemically powered turbo generators are also employed.

In addition to the batteries, power must be conditioned and distributed. Some cables are mounted external to the SRMS in structural raceways or tunnels.

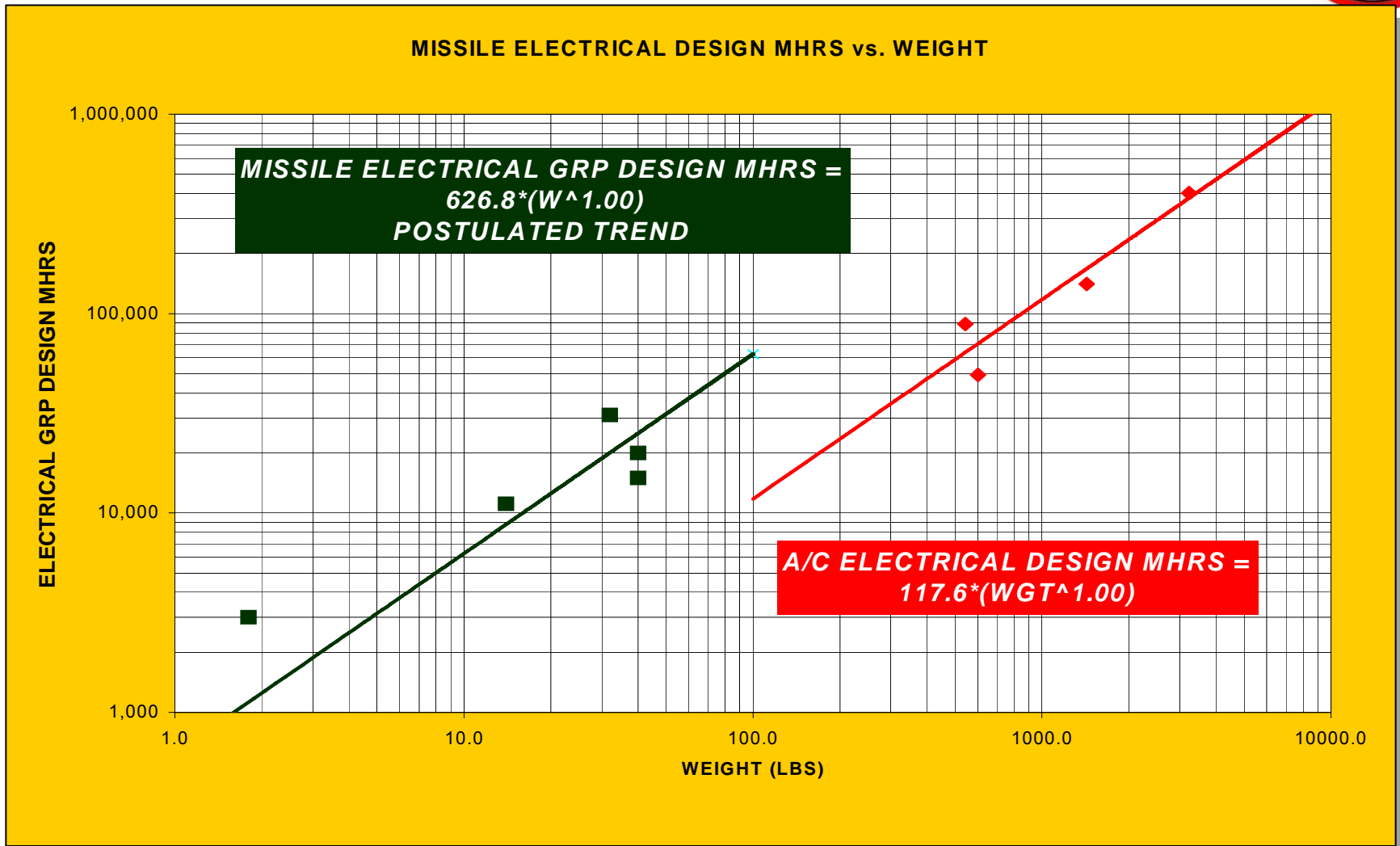


Electrical – Development Cost



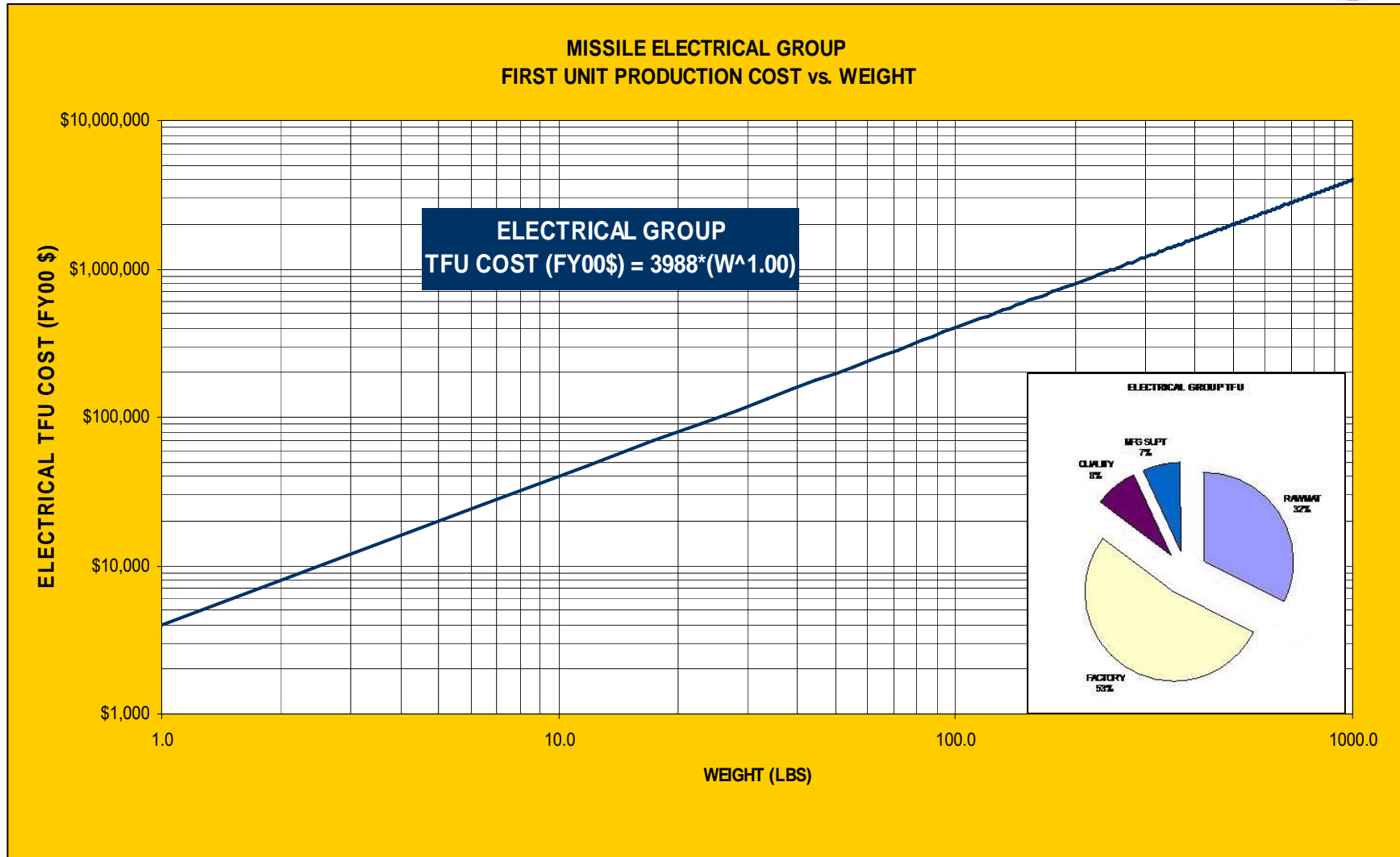


Electrical – Development Cost



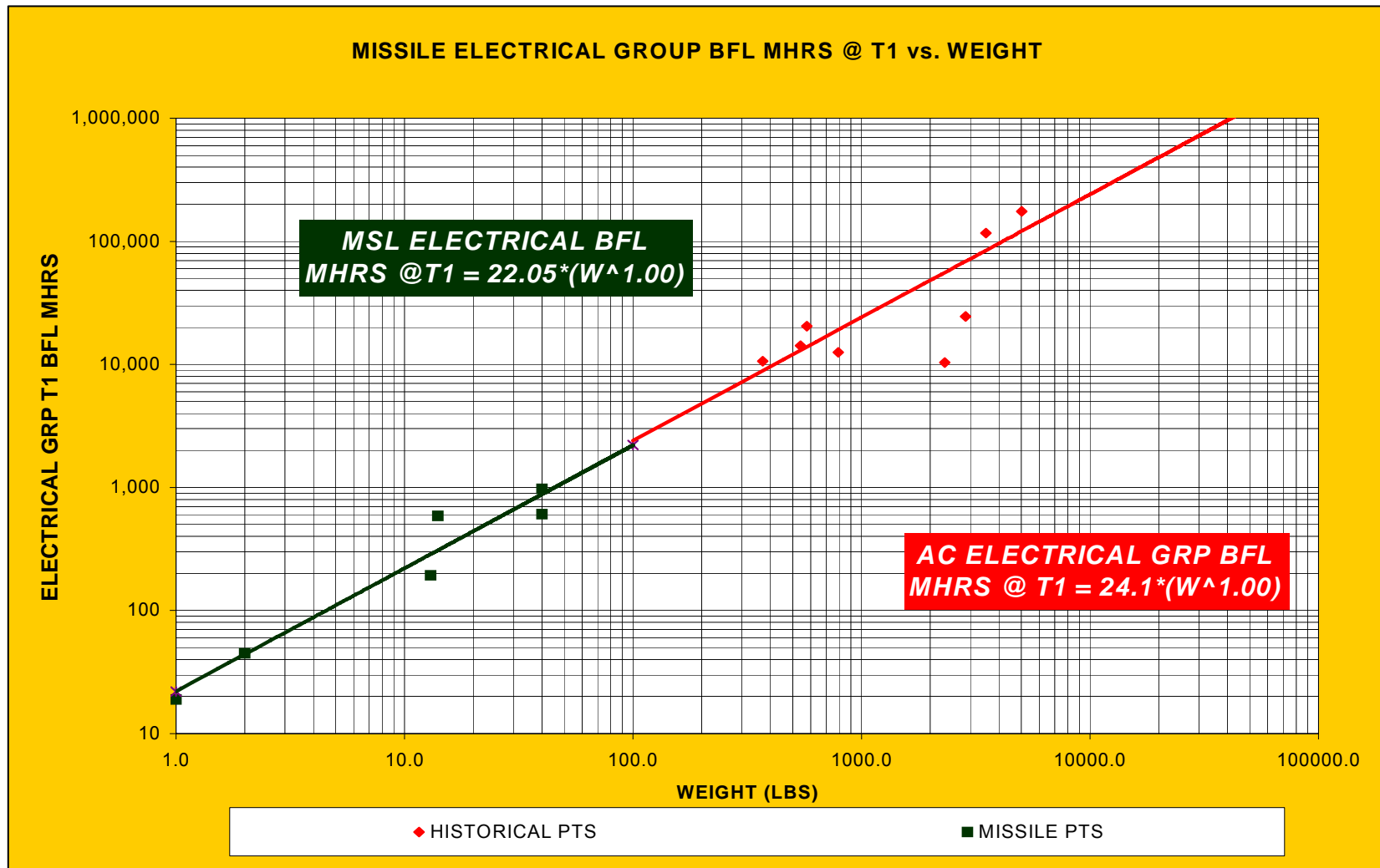


Electrical – TFU Production Cost

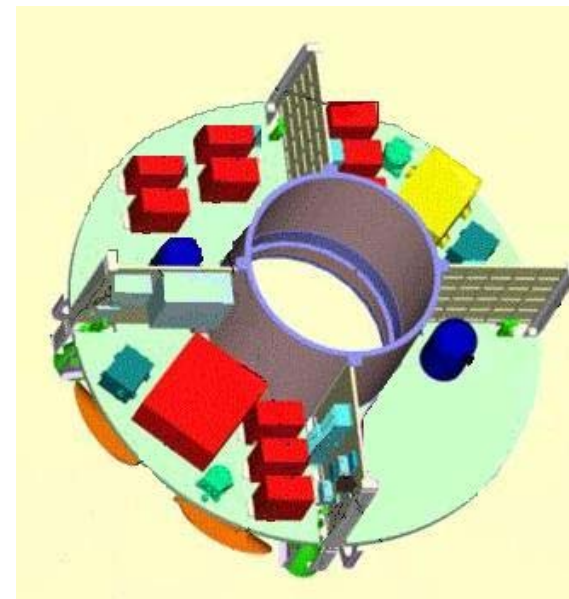




Electrical – TFU Production Cost



Guidance & Control Electronics



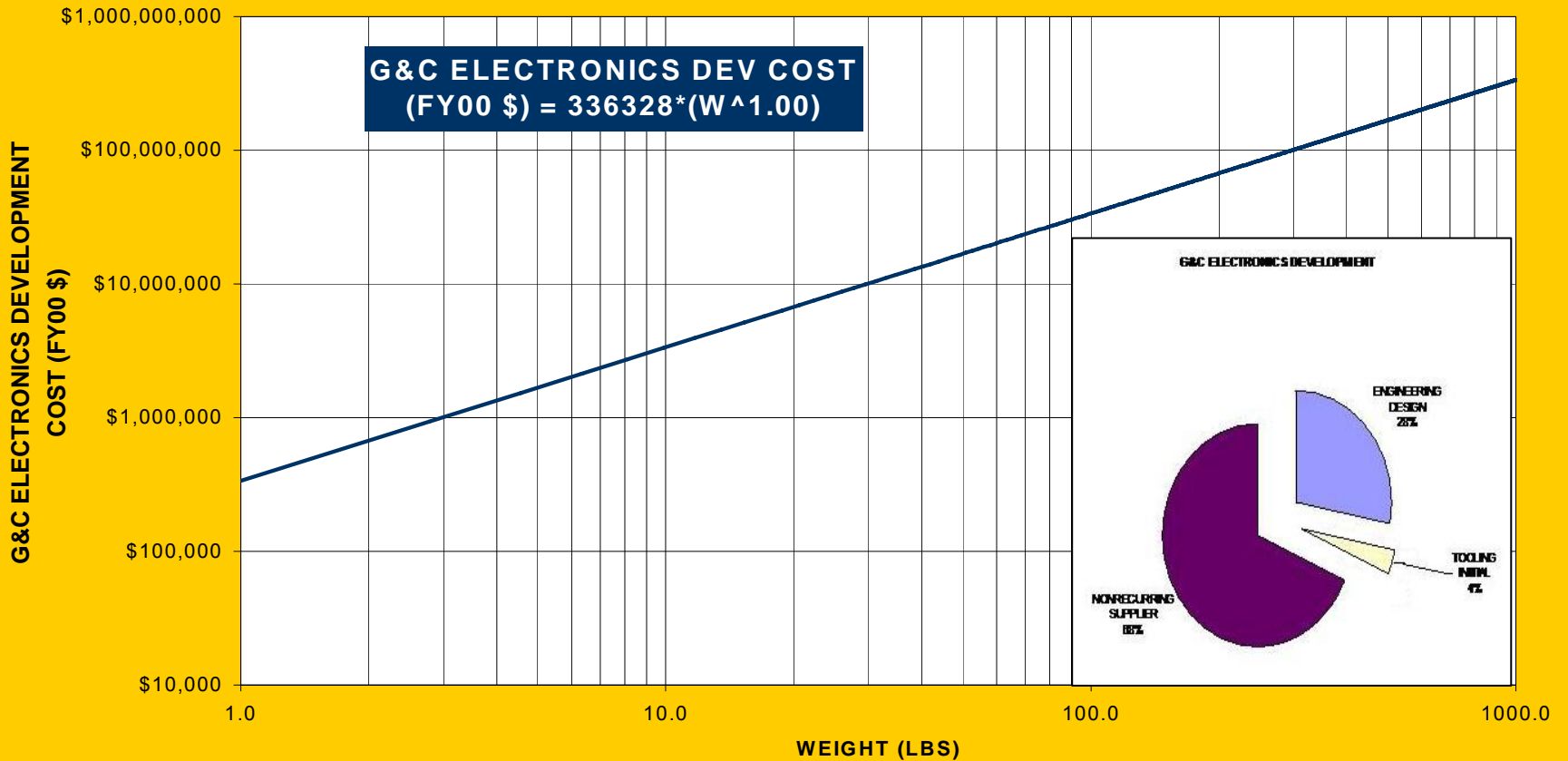
The Guidance and Control section includes Navigation systems (typically inertial and GPS, Autopilot/mission computer that steers the booster and may also include Sensors or communications links.

In addition to the electronic LRUs, provisions are required. These includes interconnect cables, mounts, and shielding as required.



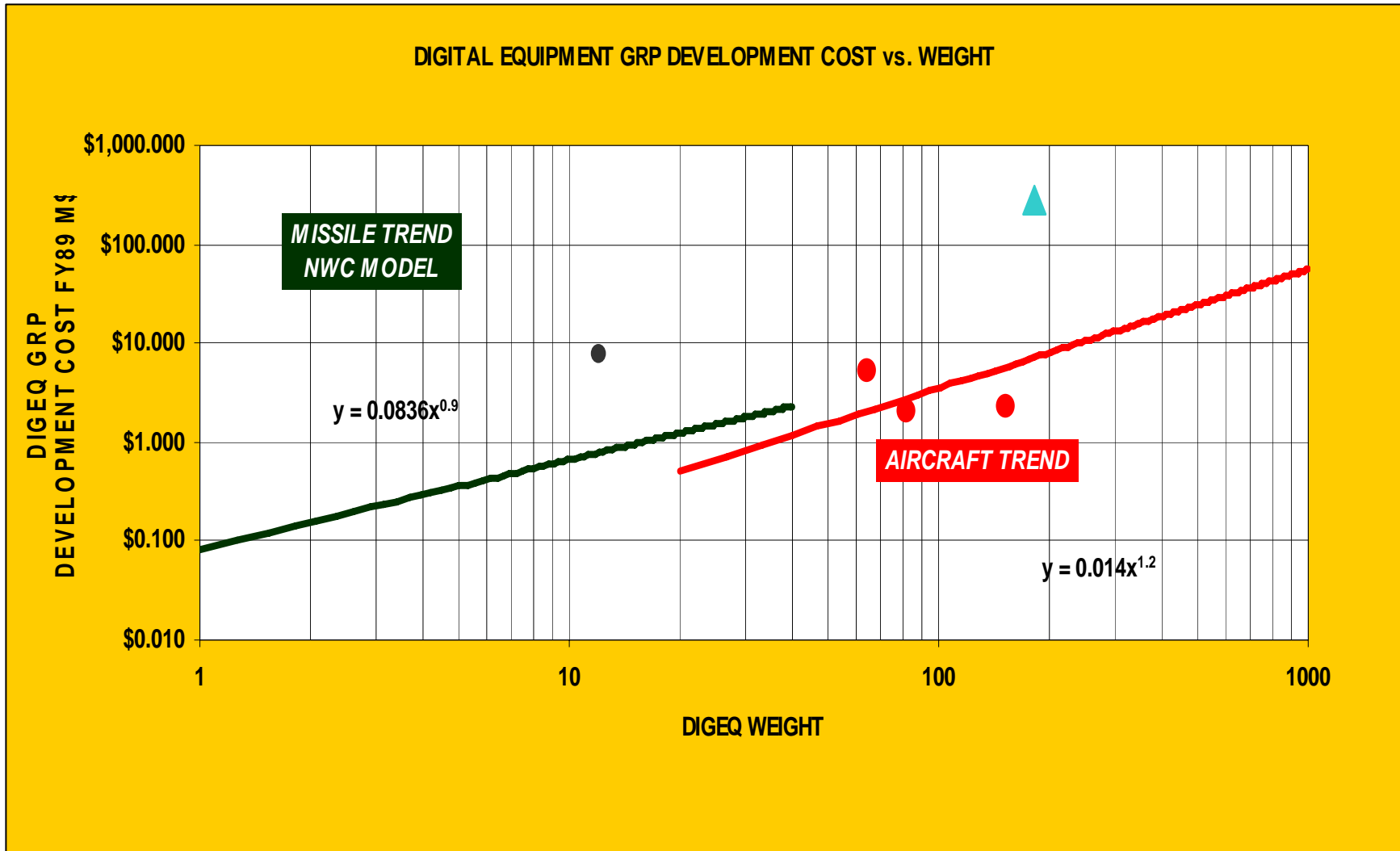
G&C Electronics – Development Cost

GUIDANCE & CONTROL ELECTRONICS DEVELOPMENT COST vs. WEIGHT



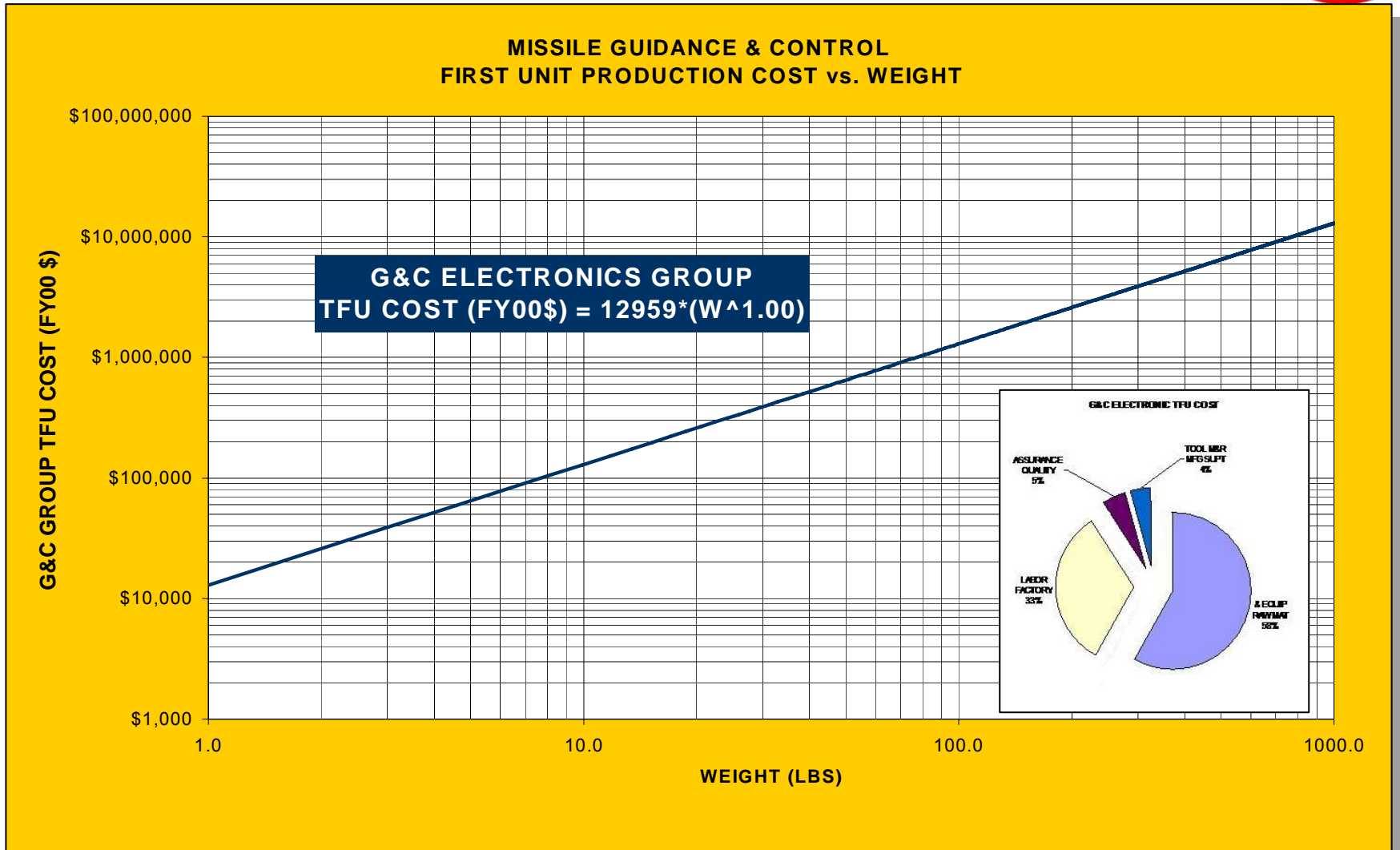


G&C Electronics – Development Cost



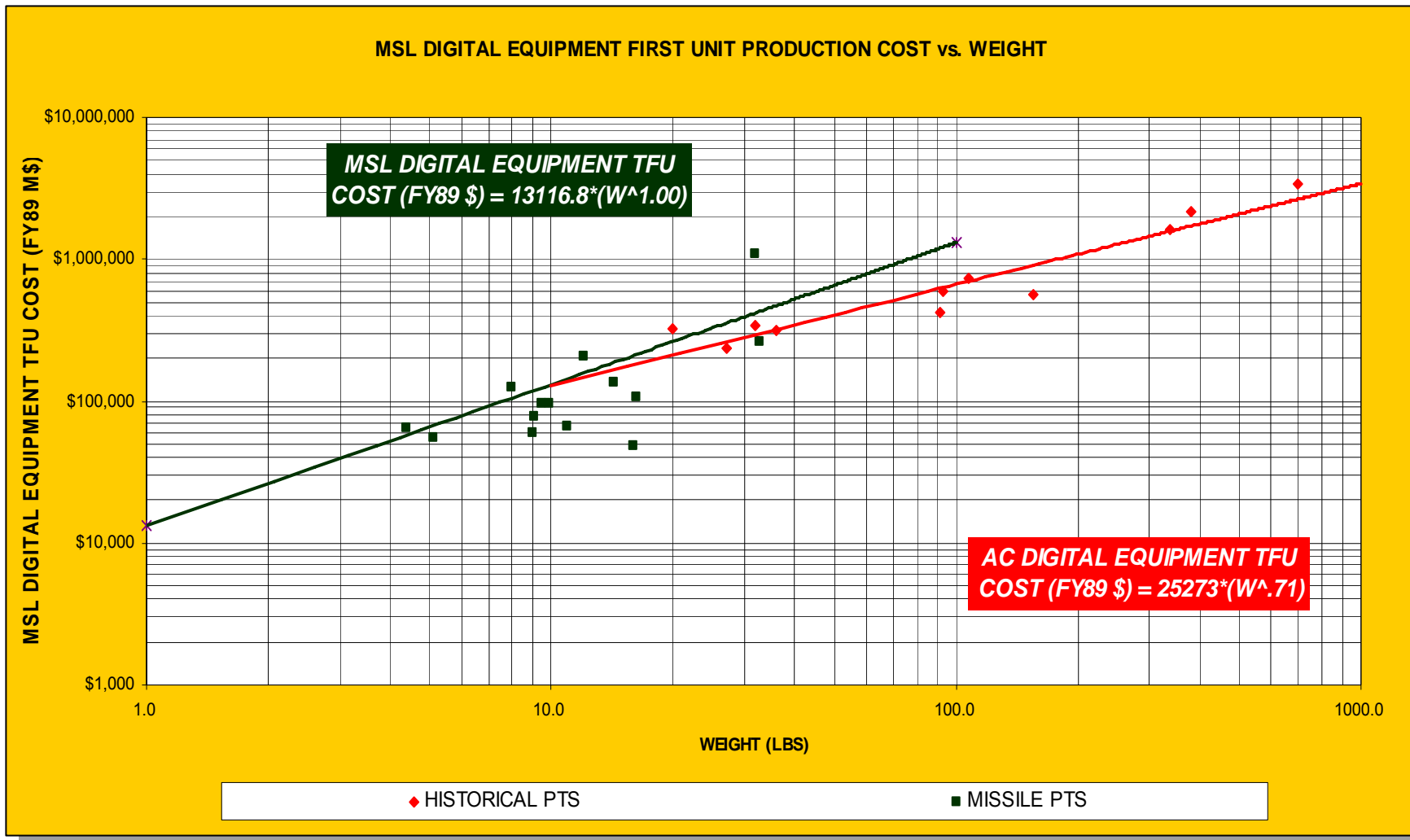


G&C Electronics – TFU Production Cost





G&C Electronics – TFU Production Cost





Guidance & Control Software Cost

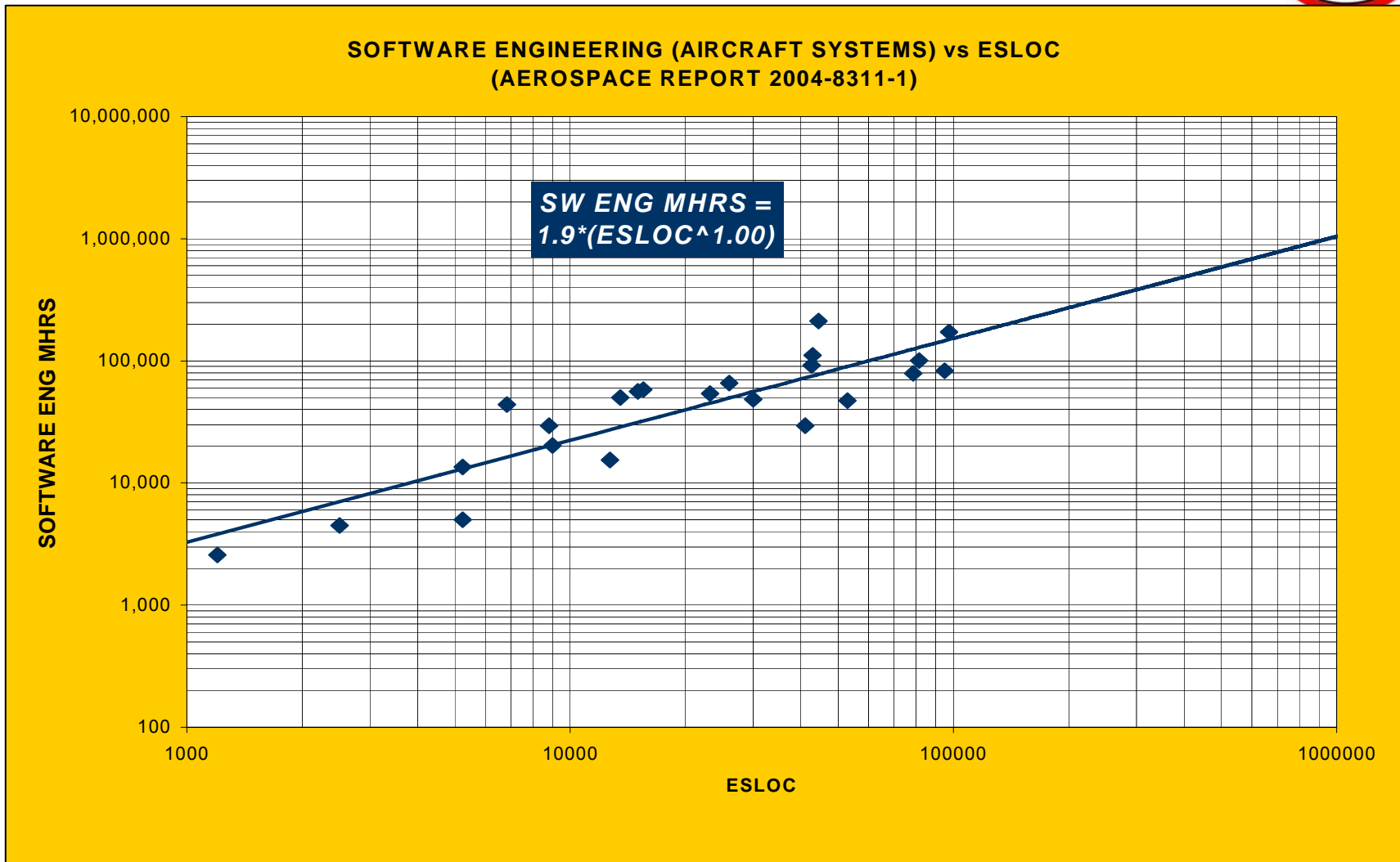
MISSILE OPERATING ENVIRONMENT

	HIGH	LOW	AVERAGE	MEDIAN	SAMPLE SIZE
	27,500	530	8,334	5,985	18
COMMAND/CONTROL					
Guidance and Control	16,000	11,500	13,750		2
GN&C					
Attitude Determination	8,000	8,000	8,000		1
OS/EXECUTIVES					
Executive Functions	1,471	1,471	1,471		1
SIGNAL PROCESSING					
Attitude Determination	27,500	530	6,354	5,400	14
Control	1,125	1,125	1,125		1
Executive Functions	645	645	645		1
Mathematical	5,200	5,200	5,200		1
Pointing	530	530	530		1
Security	19,720	19,720	19,720		1
Track/Target	1,100	1,100	1,100		1
Utilities	27,500	5,600	14,561	10,583	3
Weapons Delivery	2,550	1,200	1,875		2
	18,648	6,370	12,432	12,278	3

Software Size Estimation – Analogies from Missile Systems



Guidance & Control Software Cost





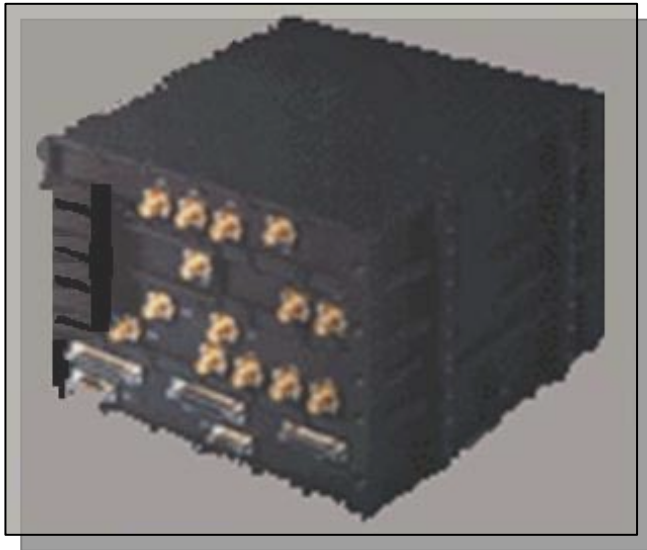
Cost Per SLOC (FY07\$)

SOFTWARE PLATFORM	COST PER		TOTAL		FY07 COST PER SLOC (WO FEE)	MEAN FY07 COST PER SLOC (WO FEE)
	ENGINEERING MHR FY2007	MHRS PER SLOC (DES/CODE/UNIT TEST)	SYSTEM ENGINEERING MHRS	ENGINEERING MHRS PER SLOC		
GROUND (LOW)	\$150	1.1	1.30	2.40	\$360	\$480
GROUND (HIGH)	\$250	1.1	1.30	2.40	\$600	
GROUND MOBILE (LOW)	\$150	1.5	1.77	3.27	\$491	\$654
GROUND MOBILE (HIGH)	\$250	1.5	1.77	3.27	\$818	
AIRBORNE (LOW)	\$150	1.9	2.24	4.14	\$621	\$828
AIRBORNE (HIGH)	\$250	1.9	2.24	4.14	\$1,036	
UNMANNED SPACE (LOW)	\$150	3.1	3.66	6.76	\$1,014	\$1,352
UNMANNED SPACE (HIGH)	\$250	3.1	3.66	6.76	\$1,690	

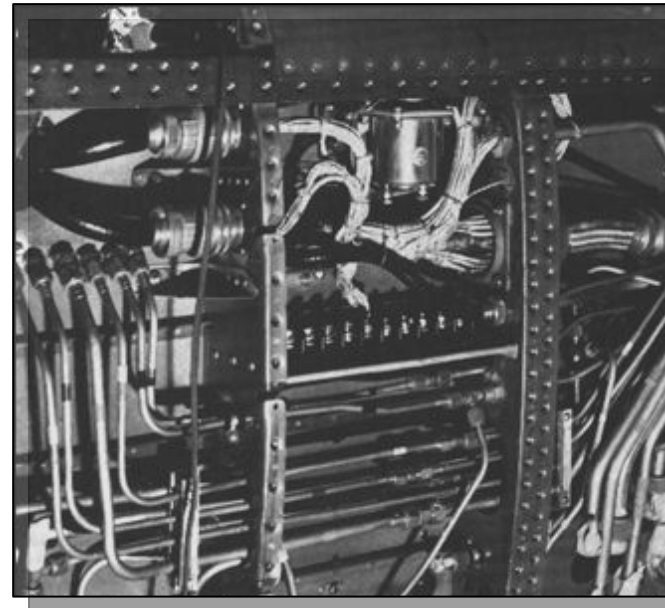
The cost of a line of code varies with contractors and by software platform or type.



Flight Termination Group



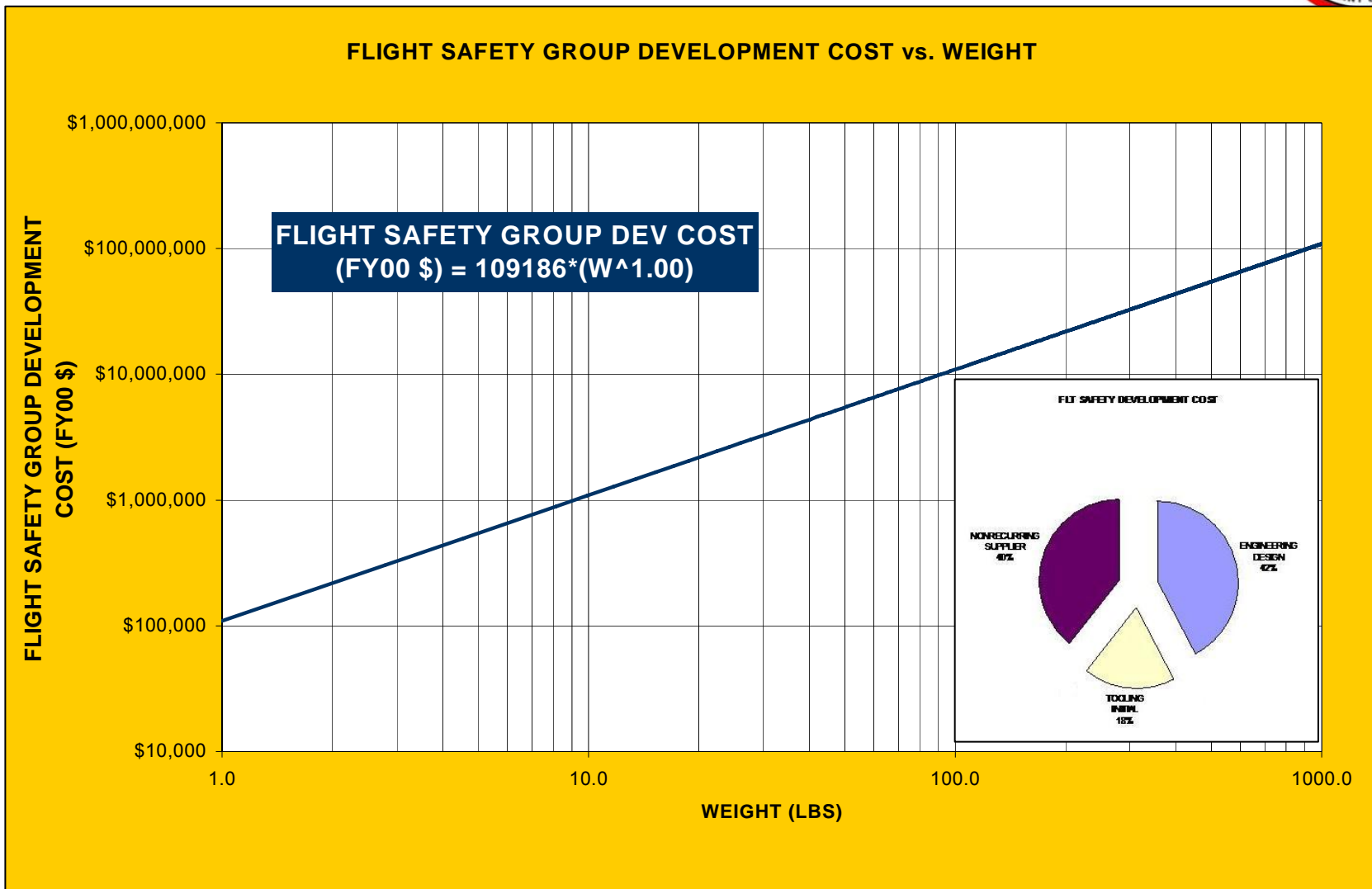
The Flight Safety group includes a communications link, an encoder, an event controller (computer and timer) and carefully positioned explosive charges.



In addition to the electronic LRUs, provisions are required. These includes interconnect cables, mounts, and shielding as required.

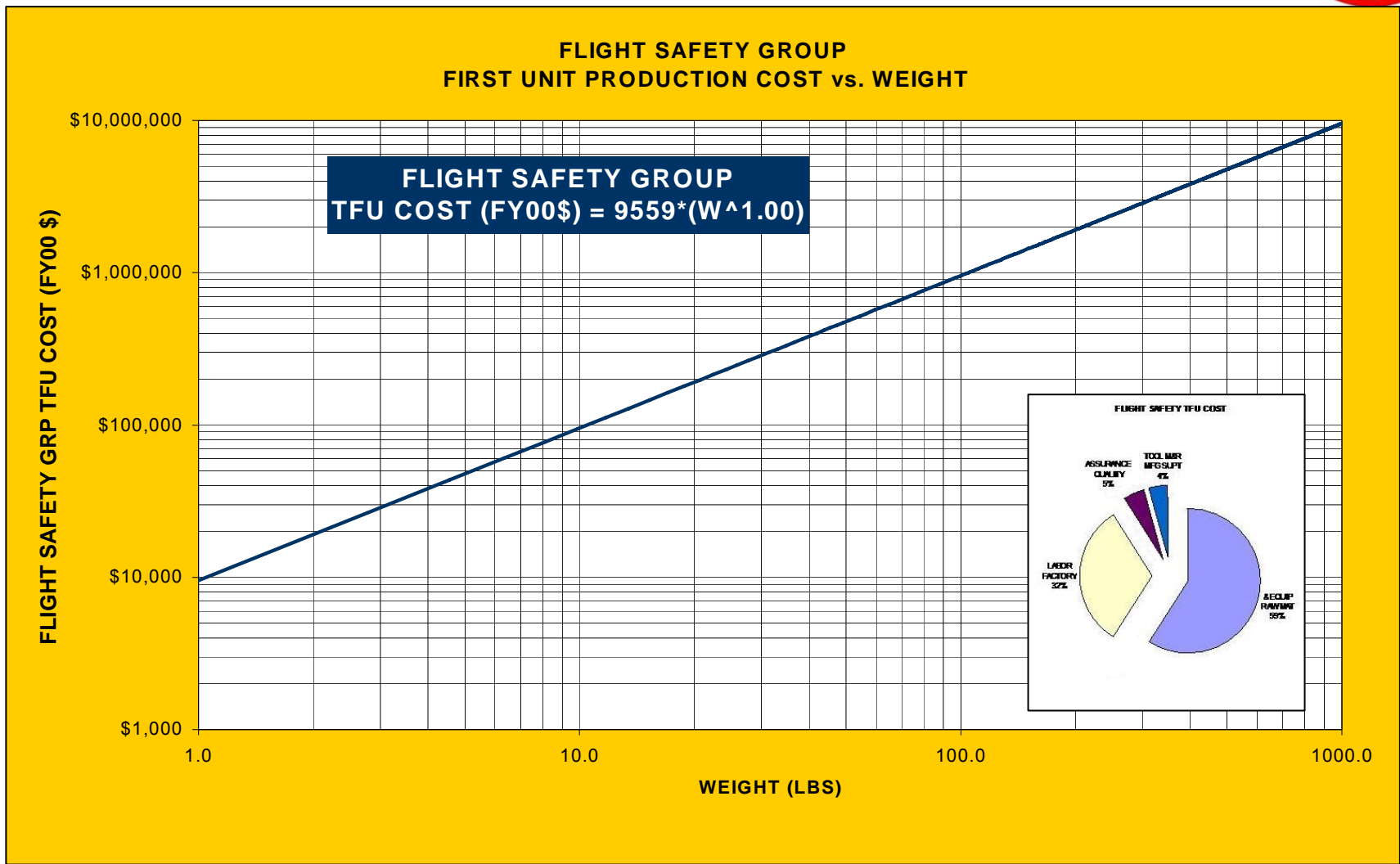


Flight Termination Group – Development Cost

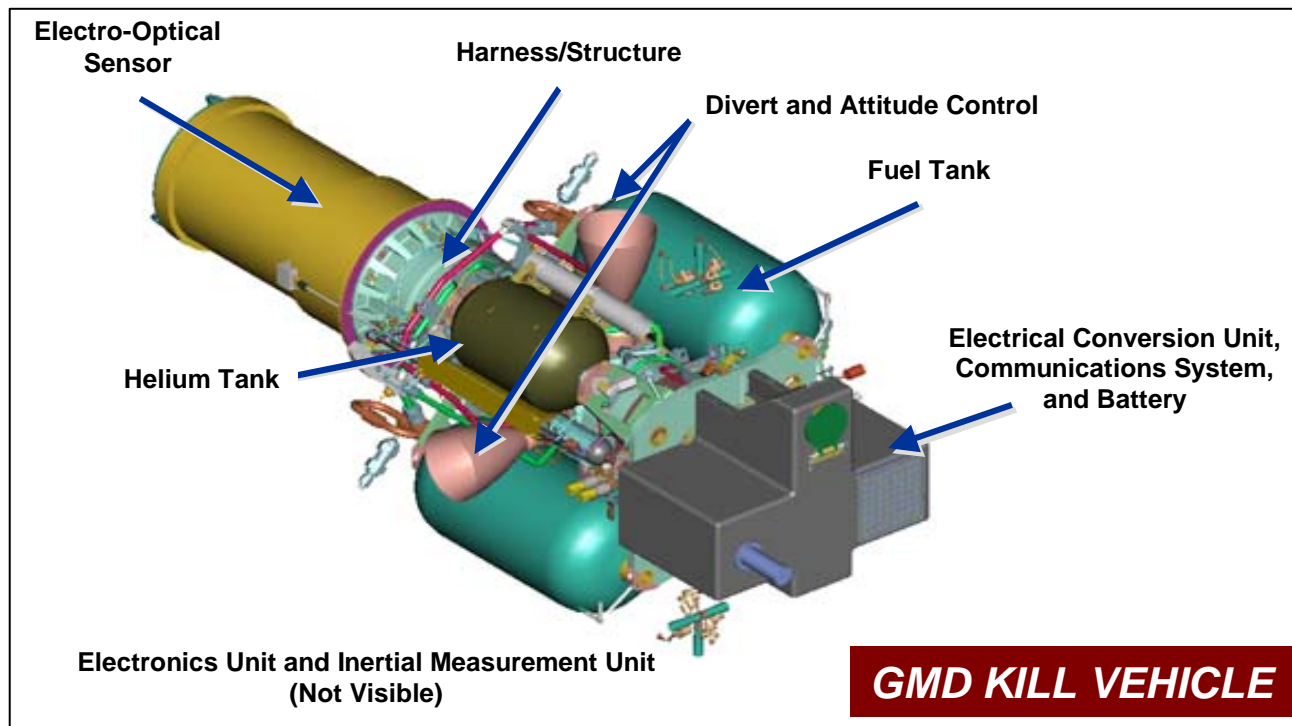
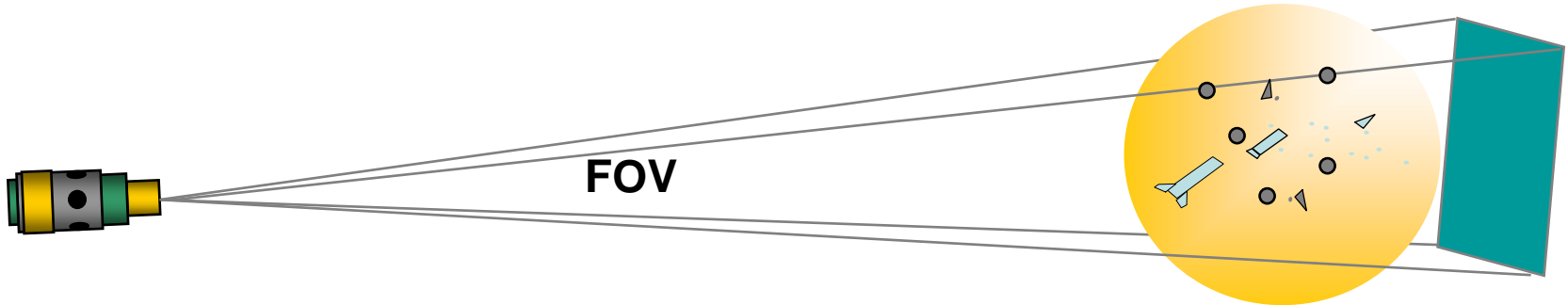




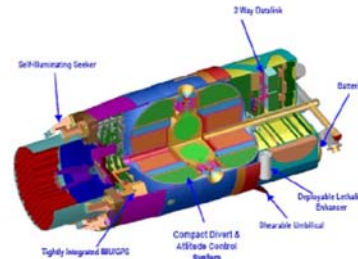
Flight Termination Group – TFU Production Cost



EKV System Technical Characteristics



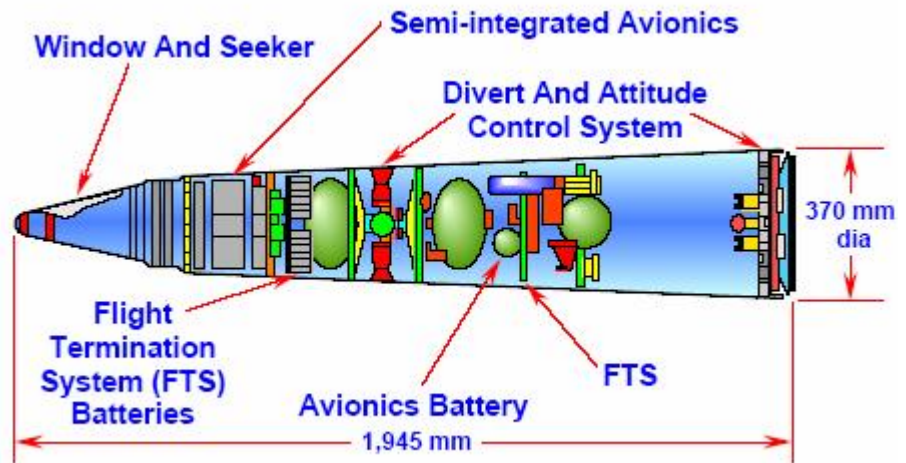
Other MDA KV Systems



MULTIPLE KILL VEHICLE

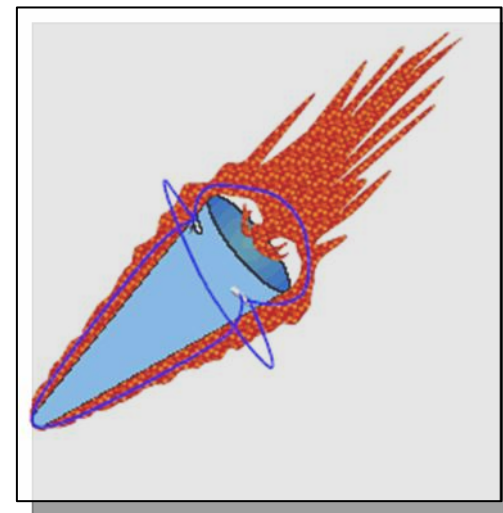
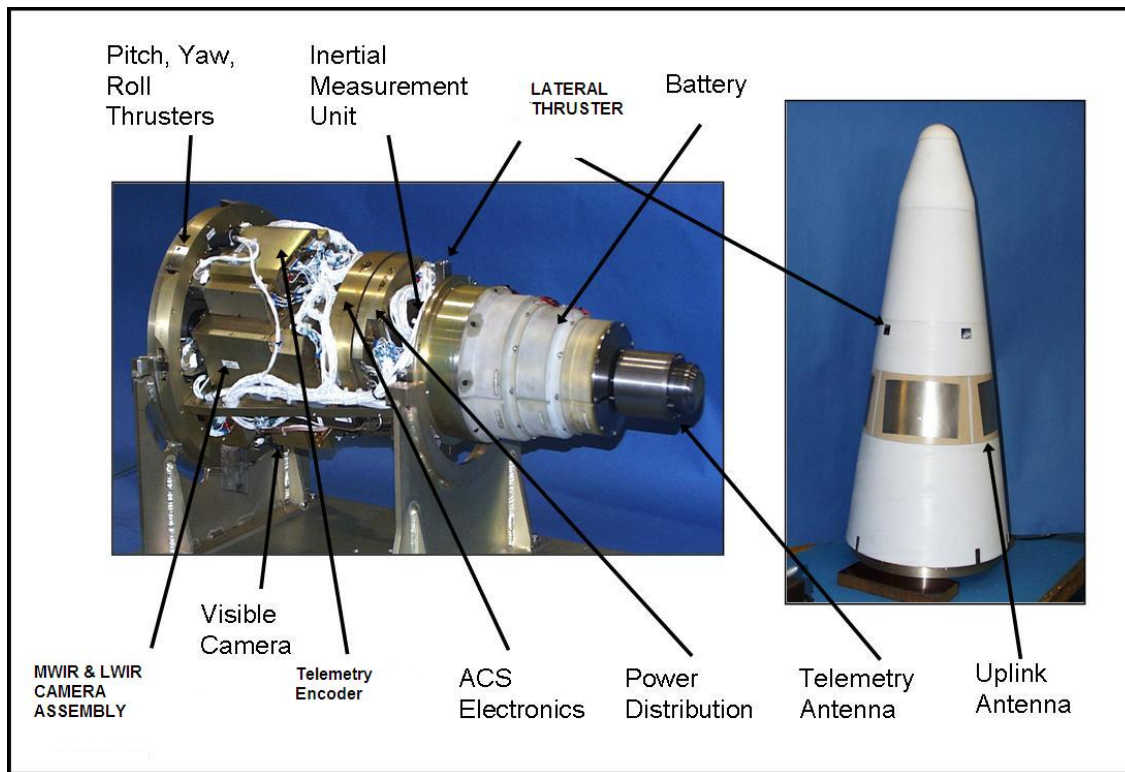


SM3 SEEKER/KV



THAAD SEEKER/KV

Target System Payloads

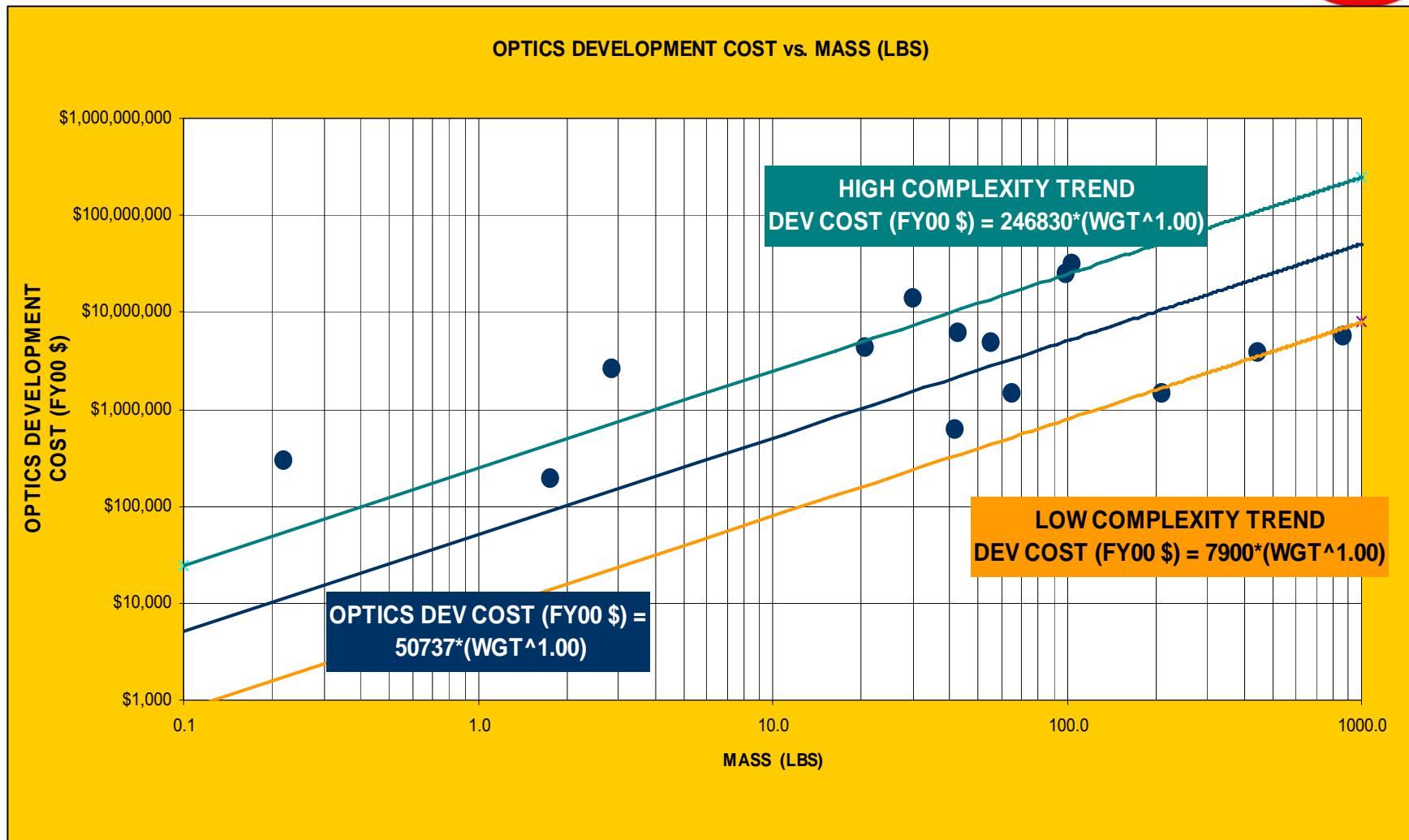


INSTRUMENTED RE-ENTRY VEHICLE

FLY ALONG SENSOR PACKAGE

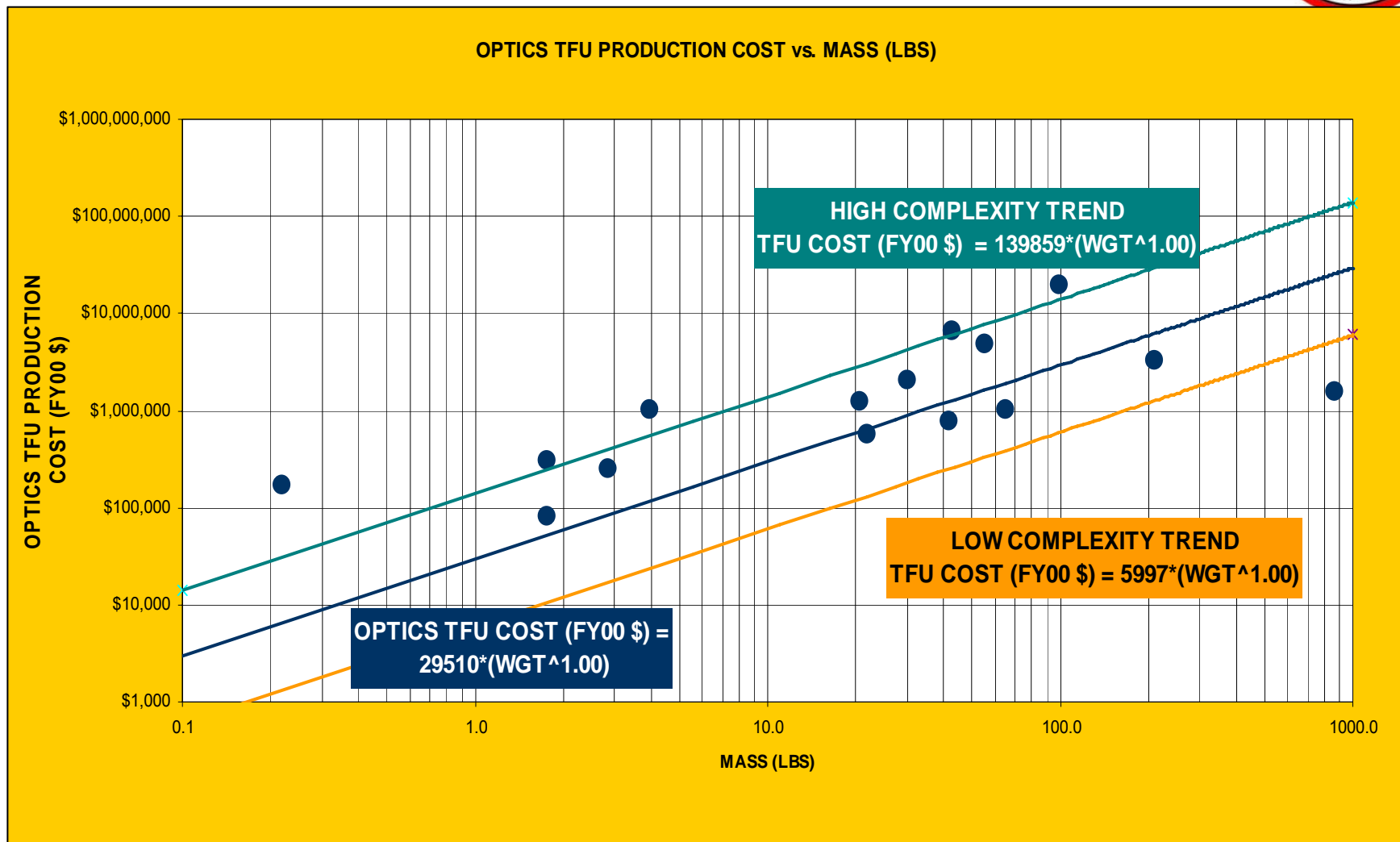


Payload CERs from Spacecraft Cost Models





Payload CERs from Spacecraft Cost Models





Booster Cost Estimating WBS Architecture

**Development
Cost**

**Production
Cost**

Prime Mission Product

- SRMs & TVC Nozzles*
- Interstage Structures*
- Electrical Power*
- Guidance & Cntl Electronics*
- G&C Software*
- Safety Systems*
- Payload Systems*

Below The Line Costs

- Ground Support Equip (Pad)*
- Ground Support Equip (Transport)*
- Data*
- Training*
- Pad Spares*
- Site Activation*



Below The Line Factors – Strategic Missiles

STRATEGIC MISSILES		
SYSTEM LEVEL WRAP FACTORS	DEVELOPMENT	PRODUCTION
SLW COST = f(PME COST)		
IACO	0.279	0.222
TEST & EVAL	0.123	0.040
SE/PM	0.349	0.250
GSE	0.032	0.137
DATA	0.020	0.020
TRAINING	0.011	0.014
SPARES	0.065	0.080
SITE ACTIVATION/ICS	0.008	0.095
OTHER LOGISTIC COMMODITIES	0.050	0.050
FACTOR TOTAL	0.937	0.908
SOURCE	SAMSO	SAMSO



Below The Line Factors – Tactical Missiles

TACTICAL MISSILES		
	DEVELOPMENT	PRODUCTION
SYSTEM LEVEL WRAP FACTORS		
SLW COST = f(PME COST)		
IACO	0.095	0.057
TEST & EVAL	0.345	0.050
SE/PM	0.590	0.150
GSE	0.040	0.048
DATA	0.079	0.026
TRAINING	0.036	0.020
SPARES	0.065	0.056
SITE ACTIVATION/ICS	0.021	0.005
OTHER LOGISTIC COMMODITIES	0.005	0.005
FACTOR TOTAL	1.276	0.417
SOURCE	NWS	BELL



Below The Line Factors – Target Missiles

TARGET MISSILES	DEVELOPMENT	PRODUCTION
SYSTEM LEVEL WRAP FACTORS		
SLW COST = f(PME COST)		
IACO	0.050	0.050
TEST & EVAL	0.070	0.000
SE/PM	0.200	0.100
GSE	0.167	0.000
DATA	0.080	0.010
TRAINING	0.050	0.000
SPARES	0.000	0.020
SITE ACTIVATION/ICS	0.000	0.050
OTHER LOGISTIC COMMODITIES	0.000	0.000
FACTOR TOTAL	0.617	0.230
SOURCE	BELL	BELL



Transporter-Erector-Launcher (TEL) Systems

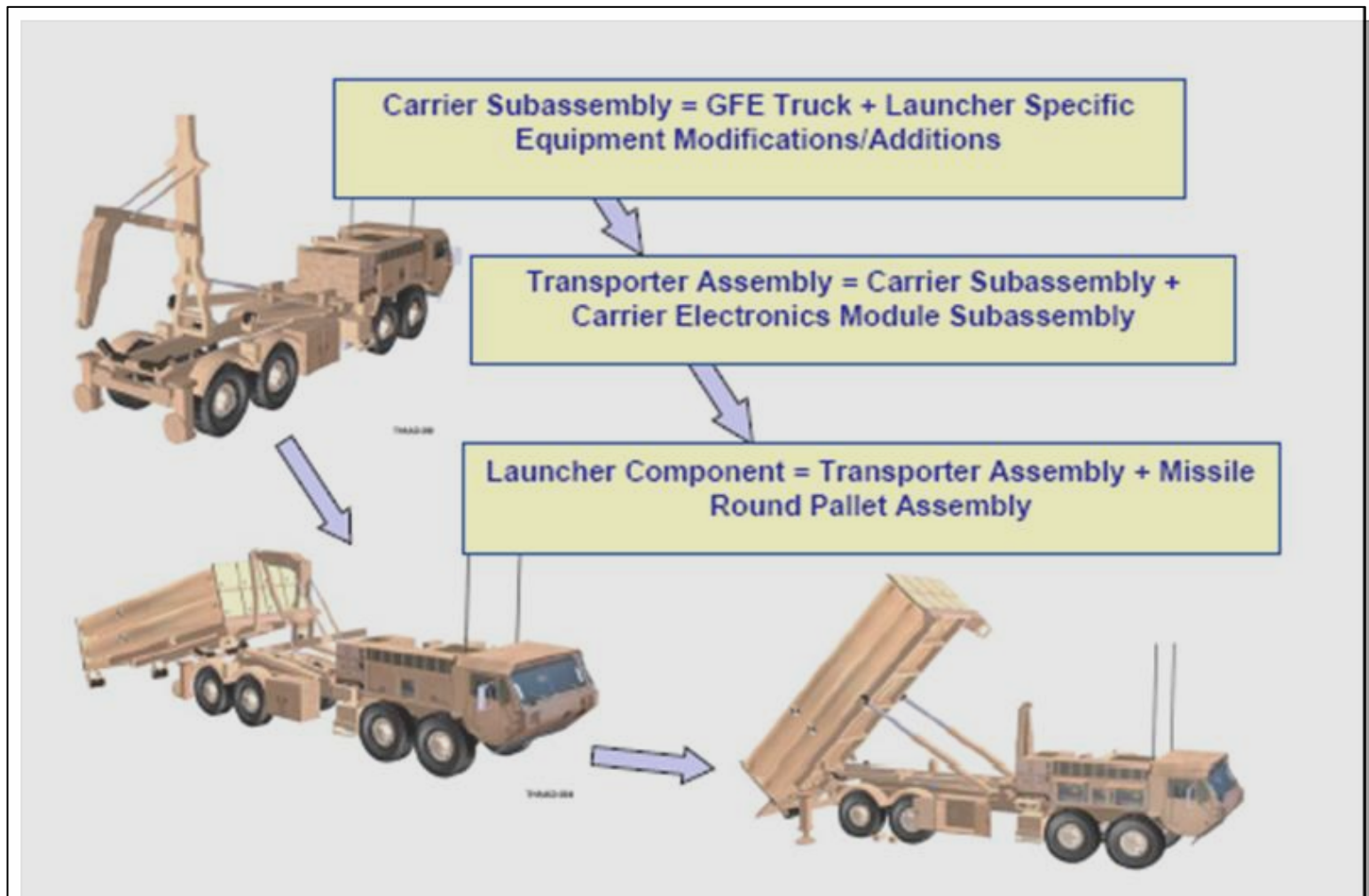


KEI TEL SYSTEM



PAC3 TEL SYSTEM

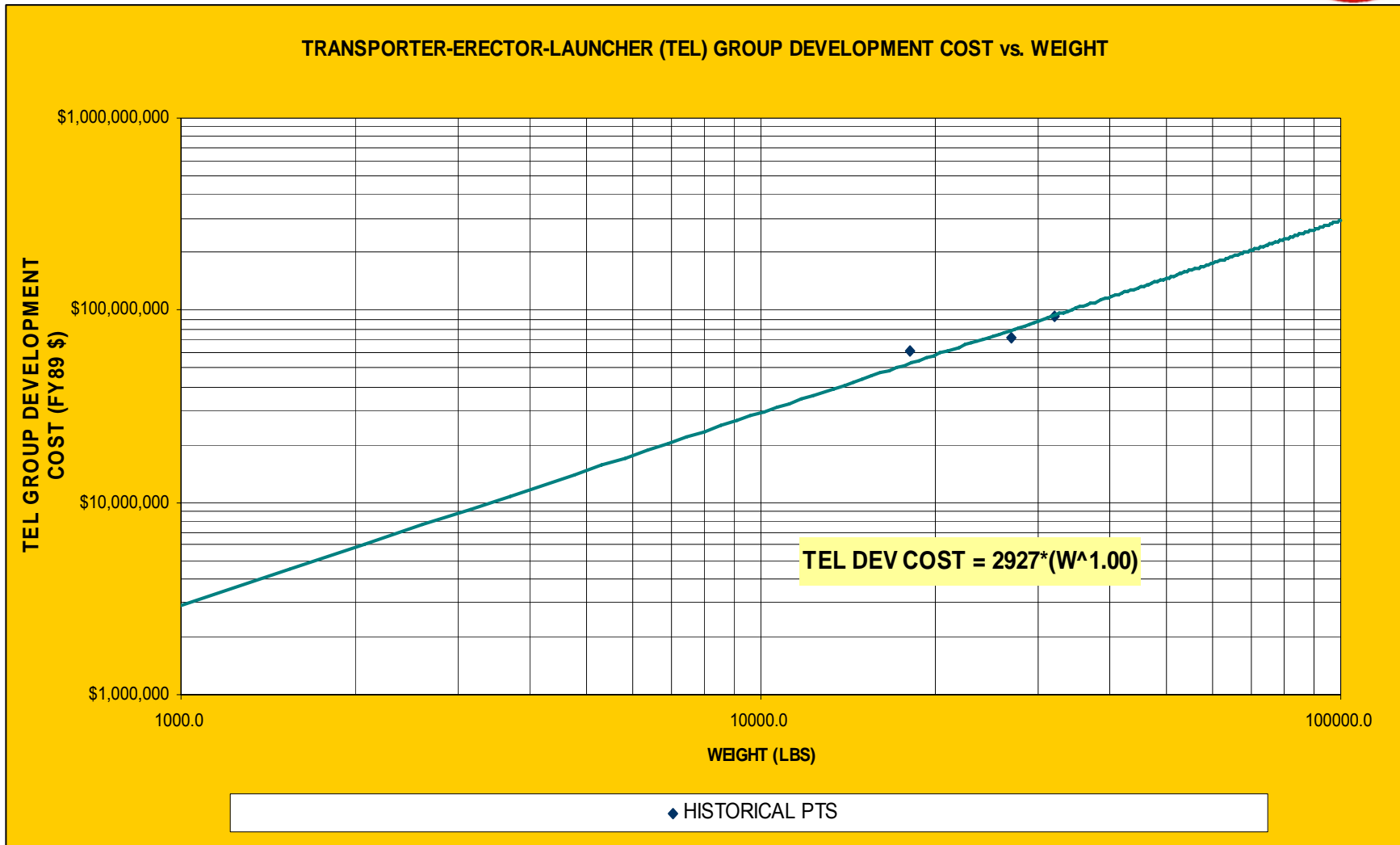
Transporter-Erector-Launcher (TEL) Systems



THAAD TEL SYSTEM

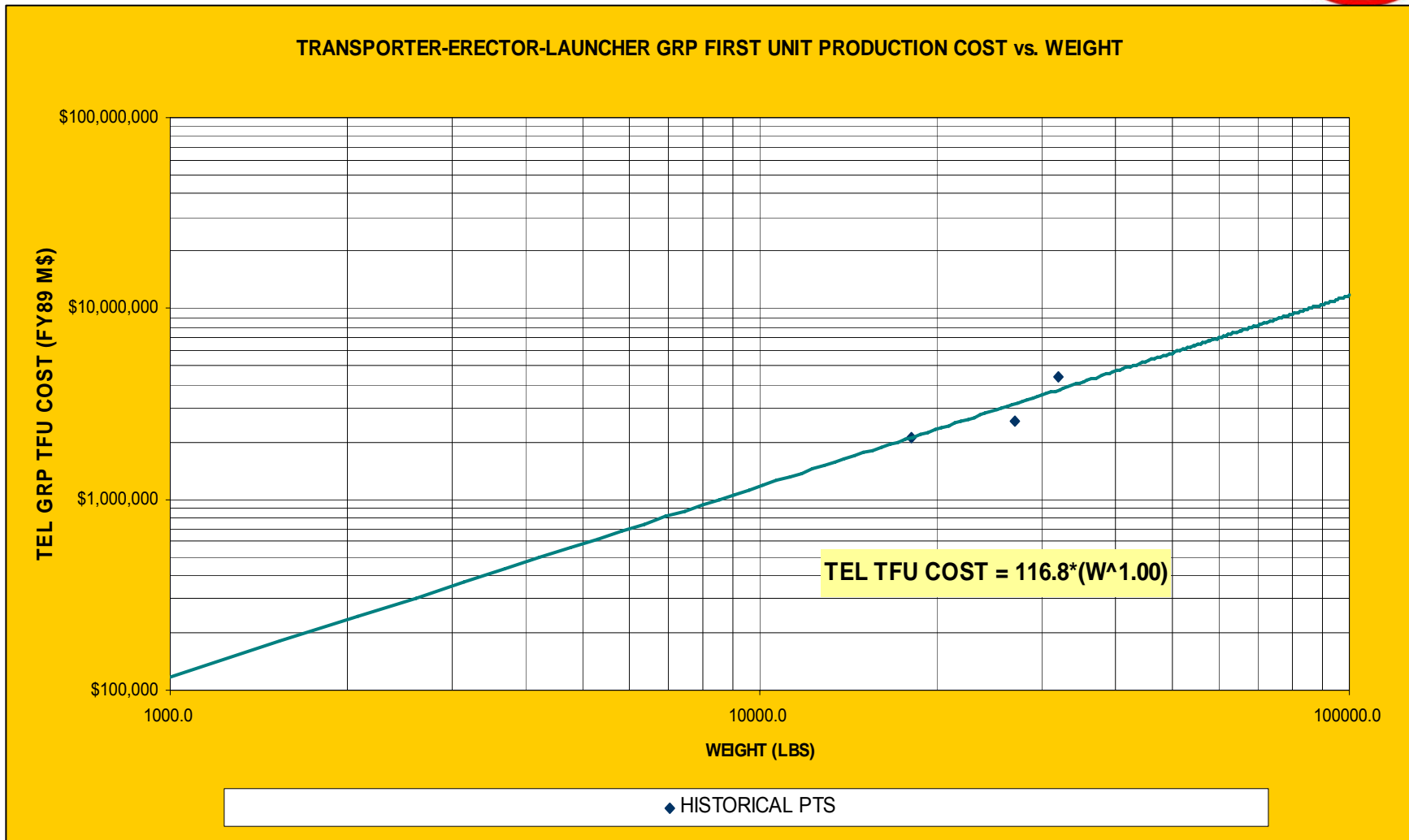


TEL Development CER





TEL TFU Production CER



SILO System for Ground-Based Interceptor

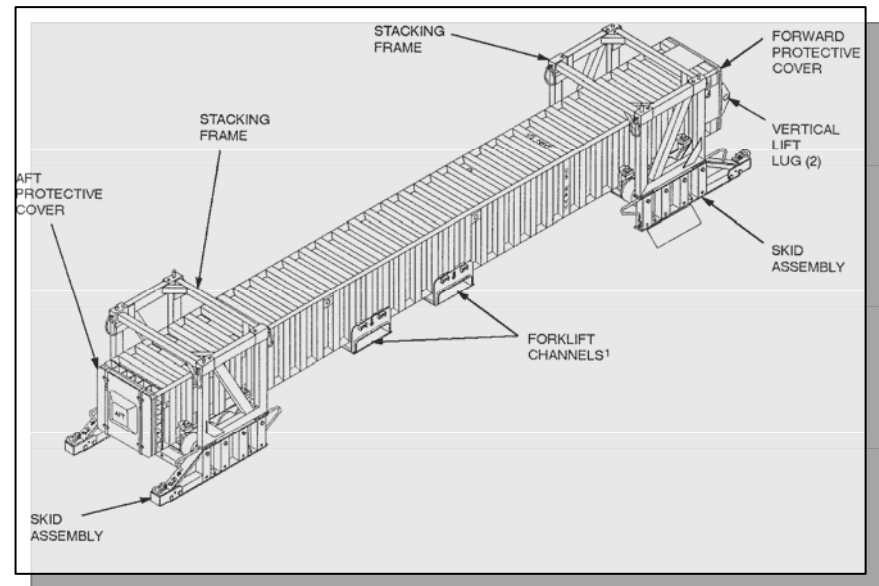
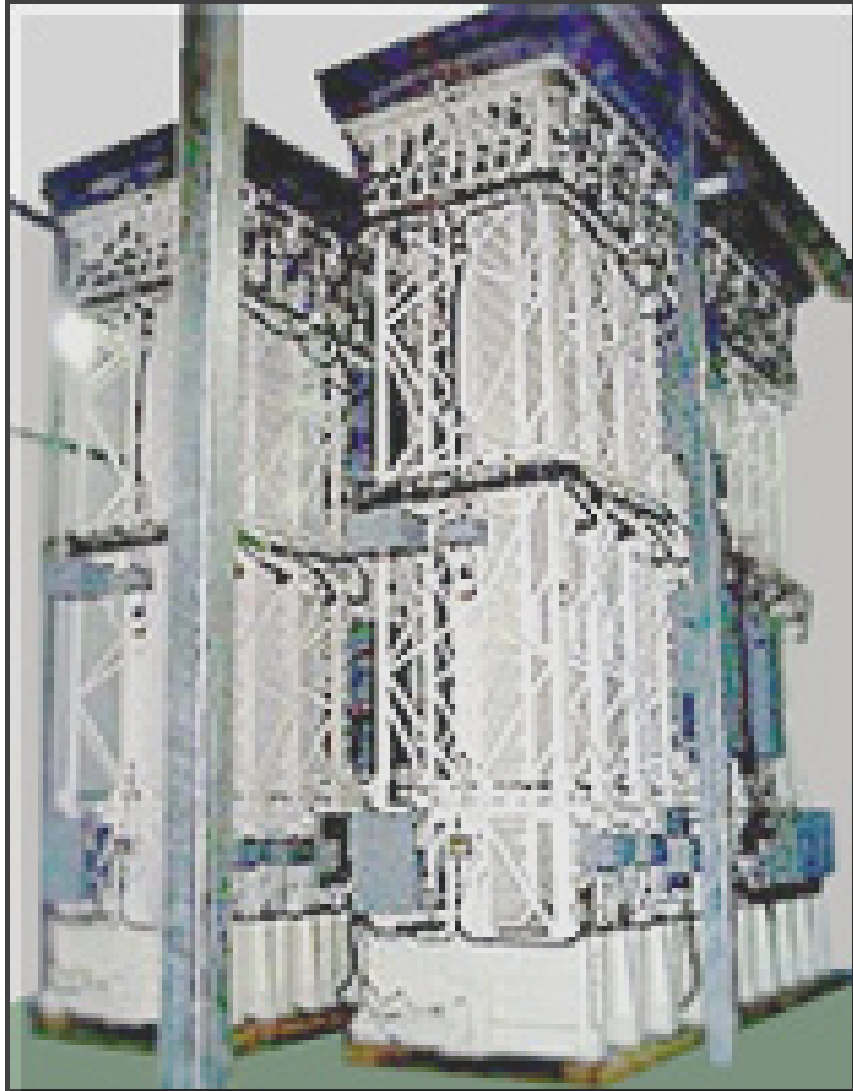


***Typical SILO arrangement for
Ground-Based Interceptor
Missile Field.***

MK41 Vertical Launch System for Sea-Based Interceptors

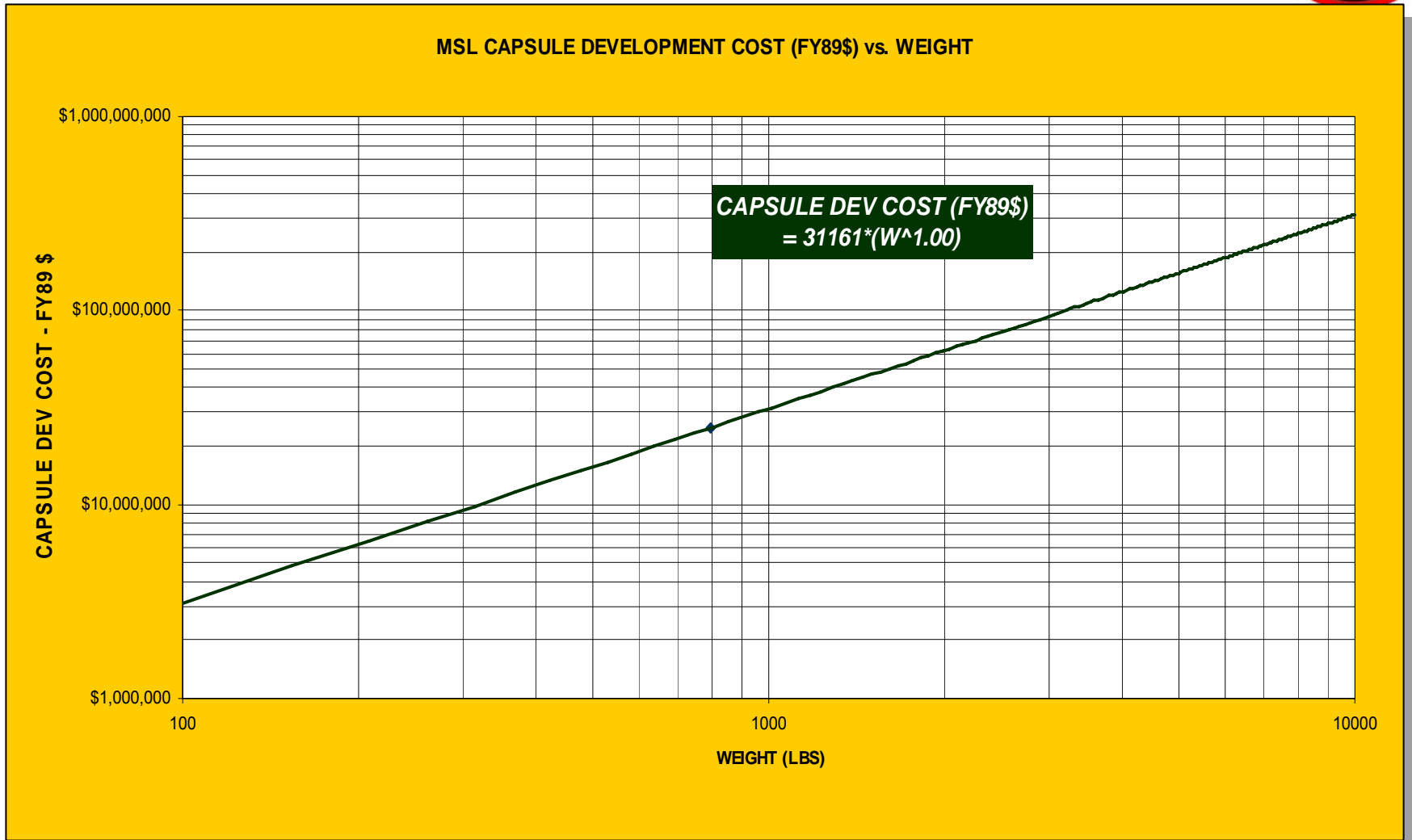


MK41 Vertical Launch System for Sea-Based Interceptors



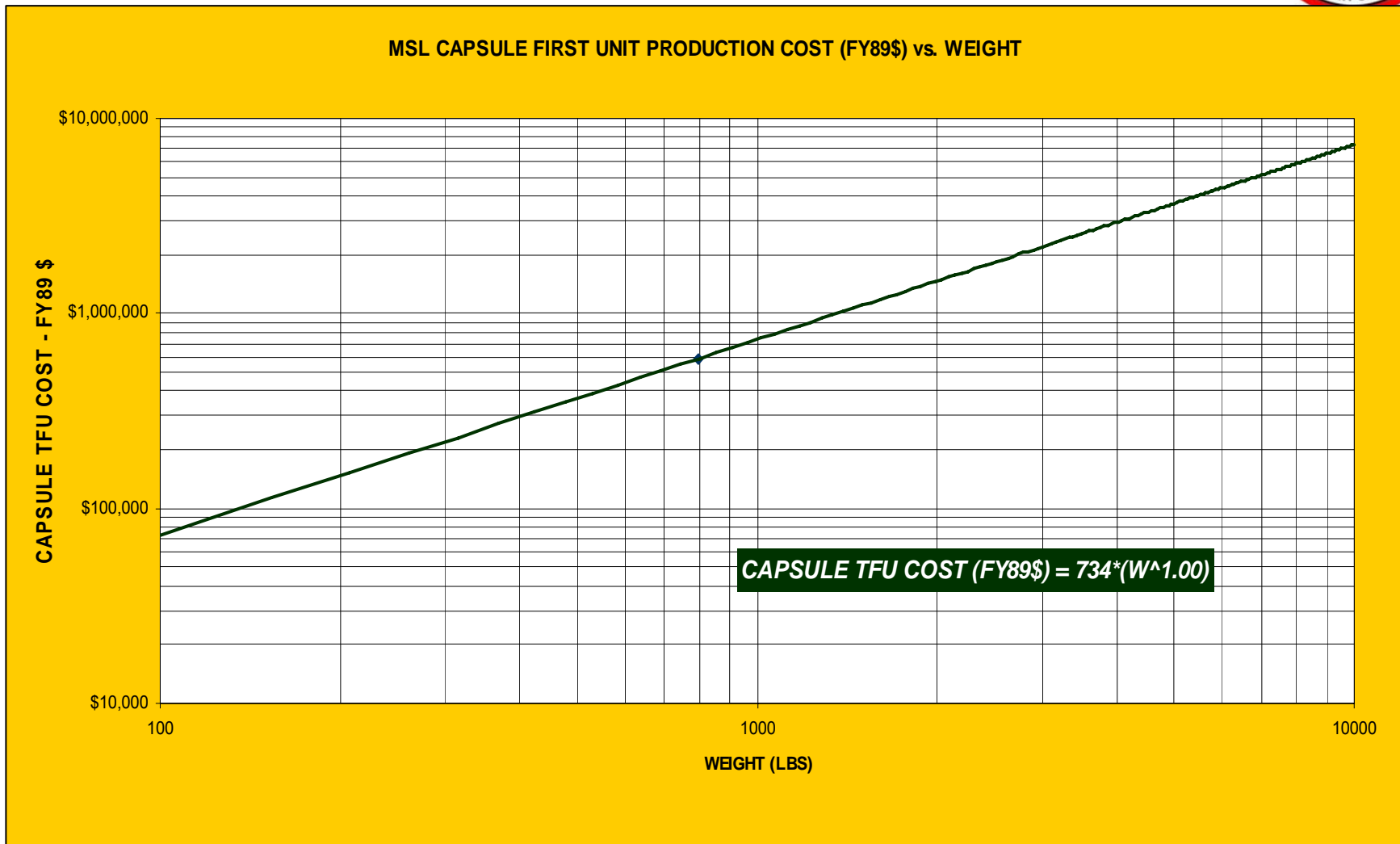


Capsule Development CER

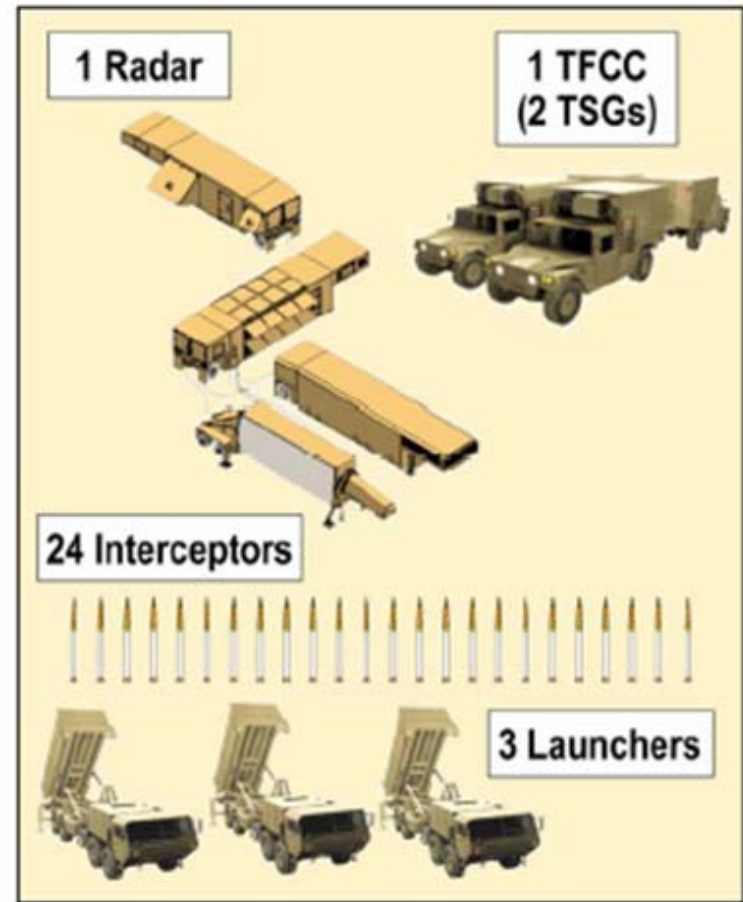
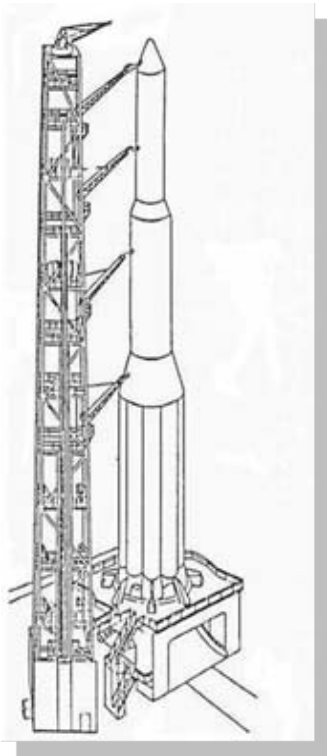




Capsule TFU Production CER



Missile System O&S Cost & Launch Operations For MDA Target Systems





ANNUAL FORCE O&S COSTS (FY06 M\$)		FOUO			S9 20-Sep-06	
NOTIONAL BALLISTIC MISSILE						
SYSTEM LIFE	20 YRS	(BATTERY) (4) UNITS ANNUAL O&S \$	(SUPPT CO) (1) UNITS ANNUAL O&S \$	(HQ UNIT) (1) UNITS ANNUAL O&S \$	TOTAL ANNUAL O&S COST (FY06 M\$)	TOTAL LIFE CYCLE O&S COST (FY06 M\$)
O&S ELEMENT						
UNIT MISSION PERSONNEL						
OFFICER ANNUAL COST		\$1.088	\$0.543	\$0.543	\$2.174	\$43.480
ENLISTED ANNUAL COST		\$6.176	\$4.429	\$1.950	\$12.555	\$251.100
CIVILIAN ANNUAL COST		\$0.000	\$0.000	\$0.000	\$0.000	\$0.000
TOTAL UNIT PERSONNEL COST		\$7.264	\$4.972	\$2.493	\$14.729	\$294.580
UNIT LEVEL CONSUMPTION						
GROUND VEHICLE POL COST		\$0.024	\$0.004	\$0.003	\$0.031	\$0.620
MISSILE PROPELLANT COST		\$0.000	\$0.000	\$0.000	\$0.000	\$0.000
MAINT SUPPLIES (VEHICLES)		\$0.024	\$0.005	\$0.007	\$0.036	\$0.720
MAINT SUPPLIES		\$0.468	\$1.320	\$0.117	\$1.905	\$38.100
OTHER UNIT OPERATING MATERIALS		\$0.000	\$0.000	\$0.000	\$0.000	\$0.000
TOTAL UNIT LEVEL CONSUMPTION COST		\$0.516	\$1.329	\$0.127	\$1.972	\$39.440
DEPOT LEVEL MAINTENANCE						
D-LVL MAINTENANCE					\$4.387	\$87.746
TOTAL DLM COST					\$4.387	\$87.746
SUSTAINING INVESTMENT						
REPLENISHMENT SPARES					\$8.155	\$163.100
REPLACEMENT GSE					\$0.425	\$8.500
CL IV MODS/SW MAINT/TECH SUPPT					\$0.497	\$9.932
ANNUAL TEST FIRINGS					\$34.908	\$698.160
SUSTAINING TECH SUPPORT					\$4.188	\$83.767
TOTAL SI COST					\$48.173	\$963.459
INSTALLATION SUPPORT PERSONNEL						
OFFICER ANNUAL ISP COST		\$0.000	\$0.000	\$0.181	\$0.181	\$3.620
ENLISTED ANNUAL ISP COST		\$0.000	\$0.000	\$1.097	\$1.097	\$21.940
CIVILIAN ANNUAL ISP COST		\$0.000	\$0.000	\$0.000	\$0.000	\$0.000
TOTAL ISP COST		\$0.000	\$0.000	\$1.278	\$1.278	\$25.560
INDIRECT PERSONNEL SUPPORT						
NONPAY INSTALLATION SUPT COST		\$1.232	\$0.864	\$0.623	\$2.719	\$54.380
NONPAY MEDICAL O&M COST		\$0.096	\$0.066	\$0.048	\$0.210	\$4.200
PCS PERSONNEL COST		\$0.752	\$0.519	\$0.388	\$1.659	\$33.180
TOTAL IPS COST		\$2.080	\$1.449	\$1.059	\$4.588	\$91.760
PERSONNEL ACQUISITION & TRAINING						
PERSONNEL ACQUISITION COST		\$0.284	\$0.181	\$0.157	\$0.622	\$12.440
LAUNCH CREW TRAINING		\$0.065	\$0.000	\$0.000	\$0.065	\$1.302
NON LAUNCH CREW TRAINING		\$0.192	\$0.177	\$0.131	\$0.500	\$10.000
TOTAL PAT COST		\$0.541	\$0.358	\$0.288	\$1.187	\$23.742
UNIT OPERATING COST (FY06 M\$)		\$10.401	\$8.108	\$5.245	\$76.314	\$1,526.287
ANNUAL OP COST PER UNIT		\$2.600	\$8.108	\$5.245		

FOUO

O&S Cost for Missile Fields or Firing Units

Inputs for O&S Cost - Missile Fields or Firing Units



	BATTERY	SUPPT CO	HQ UNIT
GROUND VEHICLES (LT)	3	4	15
VEH UTE (MILES/YR)	1250	1250	1250
LT VEH MILES/YR	3750	5000	18750
LT VEH MILES/GAL POL	12	12	12
VEHICLE POL GAL/YR (LT)	313	417	1563
GROUND VEHICLES (HVY)	10	6	0
VEH UTE (MILES/YR)	1250	1250	1250
HVY VEH MILES/YR	12500	7500	0
HVY VEH MILES/GAL POL	5	5	5
VEHICLE POL GAL/YR (HVY)	2500	1500	0
POL \$/GAL	\$2.00	\$2.00	\$2.00
PROPELLANT \$/MISSILE YR	\$0		
VEH MAINT \$/MILE	\$0.36	\$0.36	\$0.36
MAINTENANCE MAT/MMH	\$10.19	\$10.19	\$10.19
MAINTENANCE MH/MY	1440	1440	1440
TOTAL MAINTENANCE MAT \$	\$117,367	\$1,320,376	\$117,367
RESERVED OTHER MAINT \$/YR	\$0		
RESERVED OTHER MAINT \$/YR	\$0		
RESERVED OTHER MAINT \$/YR	\$0		
OTHER MAINT \$/YR	\$0	\$0	\$0

Inputs for O&S Cost - Missile Fields or Firing Units

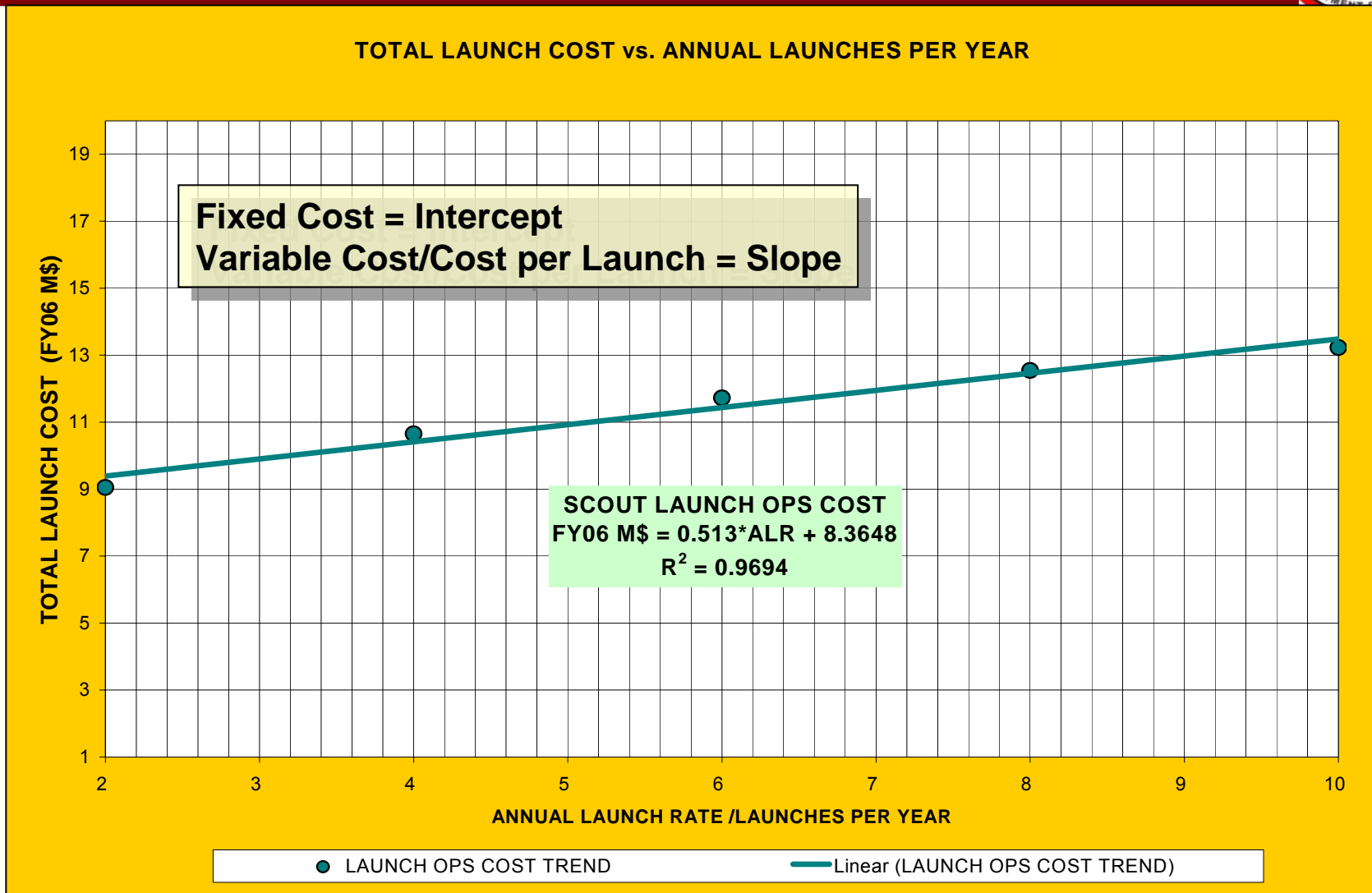


DEPOT MAINTENANCE

DEPOT MANAGEMENT	15%
DEPOT \$/MHR	\$69.86
MSL FLT SW-SLOC	75000
CMD/LAUNCH SW-SLOC	1500000
MISSILE UNIT FLYAWAY COST	\$16.500
LAUNCH SYSTEM COST/UNIT	\$8.000
FIRE UNIT ELECTRONICS COST/	\$7.500
CMD UNIT ELECTRONICS COST/	\$23.000
SYSTEM PGSE COST (M\$)	\$8.500
SYSTEM INITIAL SPARES (M\$)	\$12.000
CONDEMNATION RATE/YR	0.050
GSE REPLACEMENT RATE/YR	0.050
REPLN SPARES/MSL YR (M\$)	0.025



Target Operations: Fixed & Variable Costs





NASA Launch Ops Cost

(FY88 Cost \$ Shown)

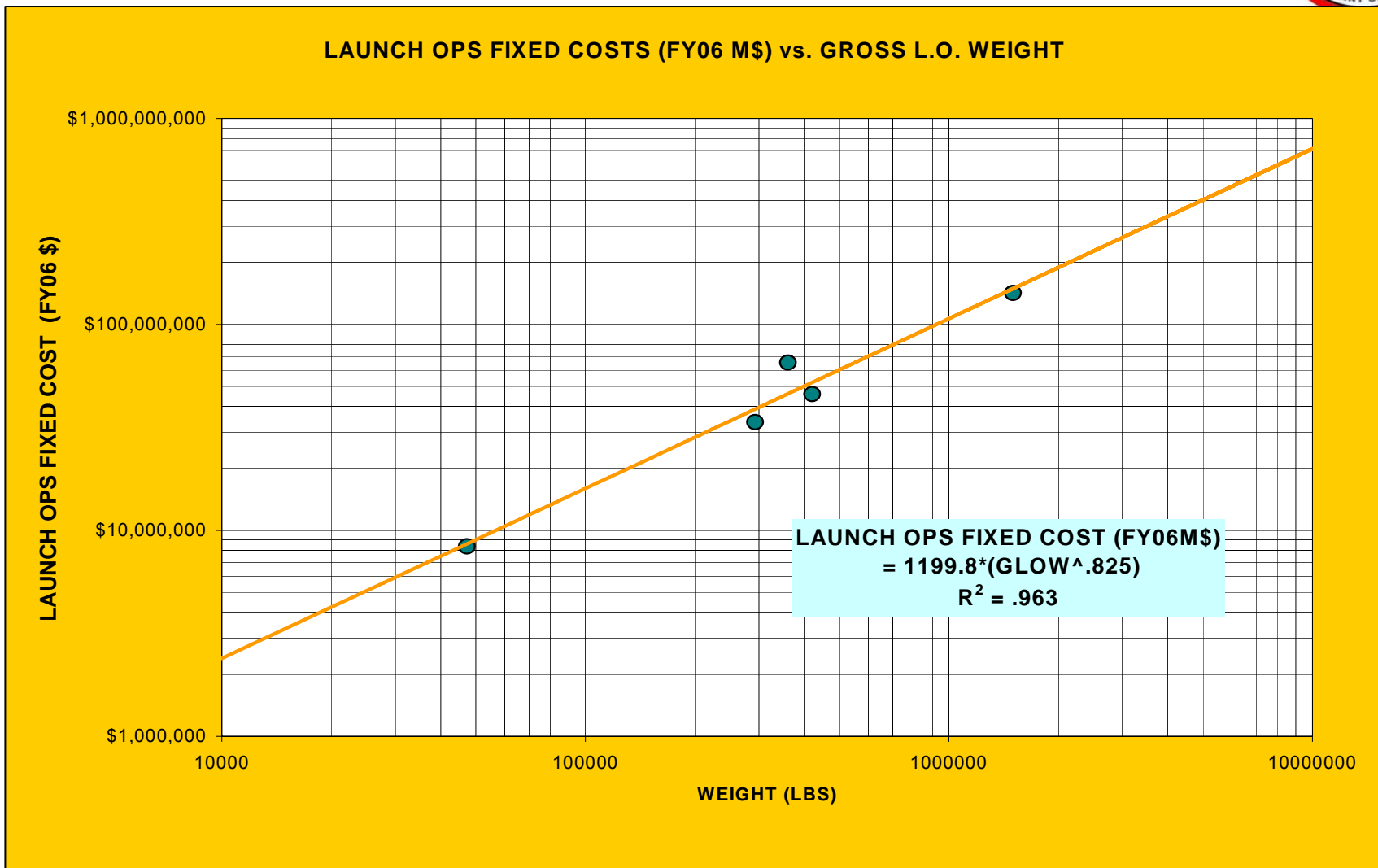


	GLOW	RATE/YR	COST TOTAL	COST/LAUNCH	FIXED COST	VARIABLE COST
TITAN 34D	1500000	2	\$73.472	\$36.736	\$63.206	\$5.133
		4	\$83.738	\$20.935		
		6	\$94.004	\$15.667		
		8	\$104.270	\$13.034		
		10	\$114.536	\$11.454		
		12	\$124.803	\$10.400		
SCOUT	47200	2	\$5.701	\$2.850	\$4.888	\$0.406
		4	\$6.513	\$1.628		
		6	\$7.326	\$1.221		
		8	\$8.138	\$1.017		
		10	\$8.951	\$0.895		
		12	\$9.763	\$0.814		
DELTA II	420500	2	\$28.702	\$14.351	\$24.662	\$2.020
		4	\$32.742	\$8.186		
		6	\$36.782	\$6.130		
		8	\$40.822	\$5.103		
		10	\$44.862	\$4.486		
		12	\$48.902	\$4.075		
ATLAS H	293000	2	\$21.977	\$10.988	\$18.877	\$1.550
		4	\$25.076	\$6.269		
		6	\$28.176	\$4.696		
		8	\$31.275	\$3.909		
		10	\$34.375	\$3.437		
		12	\$37.474	\$3.123		
ATLAS CENTAUR	360600	2	\$25.621	\$12.810	\$22.011	\$1.805
		4	\$29.230	\$7.307		
		6	\$32.839	\$5.473		
		8	\$36.449	\$4.556		
		10	\$40.058	\$4.006		
		12	\$43.667	\$3.639		

Fixed Cost = Intercept
Variable Cost/Cost per Launch = Slope

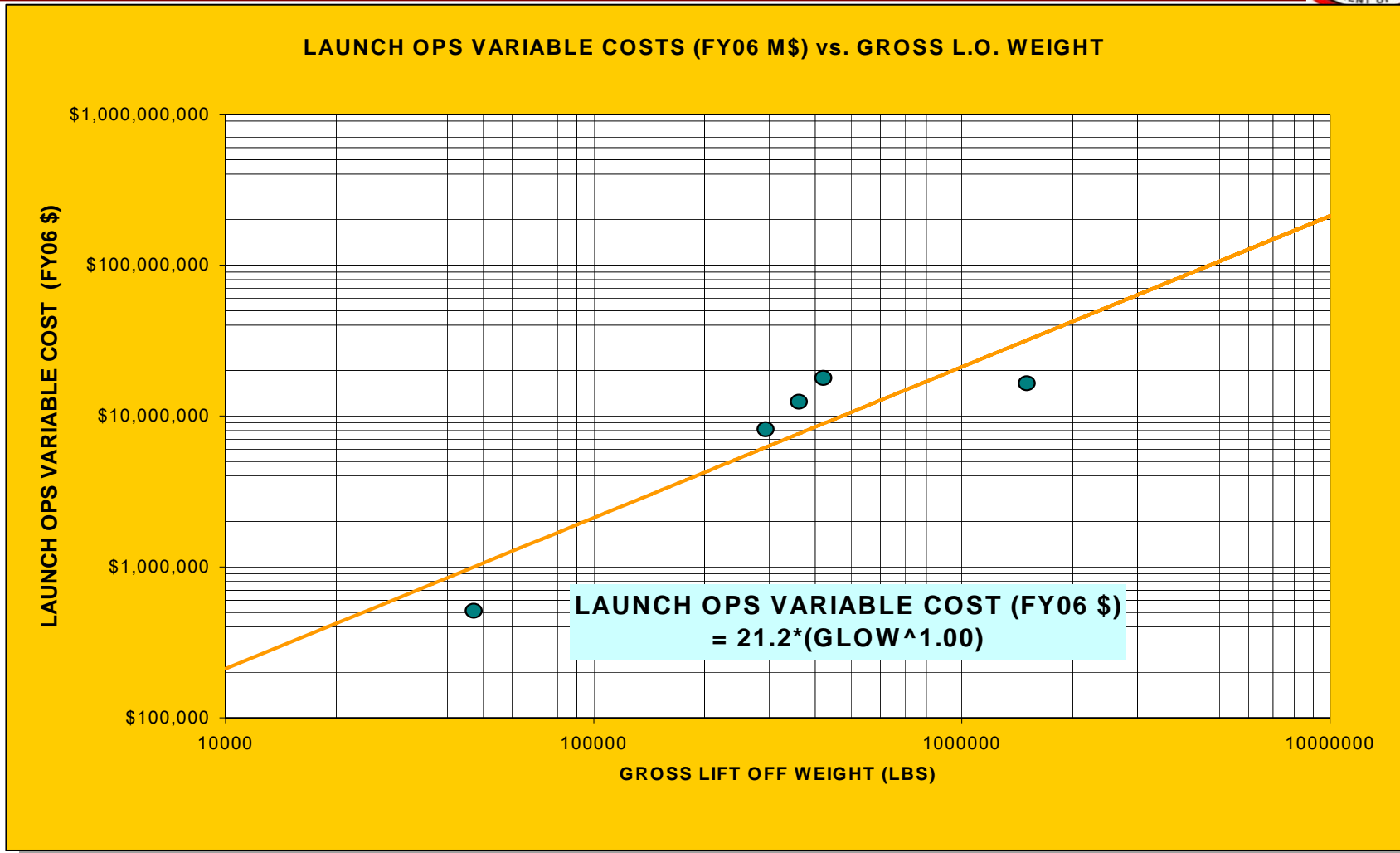


Fixed Launch Ops Cost = f(GLOW)





Variable Launch Ops Cost = f(GLOW)





Target Launch Ops Cost Model

TARGET SYSTEM:	FTF LV2
TARGET GLOW (LBS):	66000
ANNUAL LAUNCH RATE	10
FIXED LOPS COST (FY06 \$)	\$11,356,189
VARIABLE LOPS COST/LAUNCH (FY06 \$)	\$1,399,200
TOTAL TARGET SYSTEM LOPS COST	\$25,348,189
COST PER LAUNCH IN YR OF INTEREST	\$2,534,819

LAUNCH & RANGE OPS COST MODEL
 BY06 OUTLAYS AS FUNCTION OF GLOW, LAUNCHES PER YEAR
 BASED UPON NASA LAUNCH EXPERIENCE (SSCAG HNDBK)

LAUNCH COST WBS APPORTIONMENT

		FIXED FRACTION	VARIABLE FRACTION	FIXED COST ALLOCATION	VARIABLE COST ALLOCATION	ALLOCATED COST PER LAUNCH
LAUNCH SITE/RANGE COSTS						
LCC SEPM & SUPPORT	FIXED	14.85%		\$1,686,915		\$168,692
RANGE SAFETY	FIXED	34.16%		\$3,879,540		\$387,954
RANGE SECURITY OPS	FIXED	2.97%		\$337,748		\$33,775
RANGE OPS	VARIABLE		87.44%		\$12,234,421	\$1,223,442
LCC MODS	FIXED	1.45%		\$164,310		\$16,431
LAUNCH STAND MODS	FIXED	1.82%		\$207,213		\$20,721
RANGE MODS	FIXED	0.40%		\$45,642		\$4,564
TARGET VEH MODS	VARIABLE		1.11%		\$155,249	\$15,525
OTHER MODS	VARIABLE		0.00%		\$0	\$0
RANGE HEADQUARTERS SITE COSTS						
RANGE SEPM & SUPPORT	FIXED	44.34%		\$5,034,821		\$503,482
DATA REDUCTION	VARIABLE		5.30%		\$742,170	\$74,217
TARGET OFFLOAD & TRANSPORT	VARIABLE		0.28%		\$39,807	\$3,981
TARGET BUILDUP	VARIABLE		3.08%		\$431,194	\$43,119
TARGET HORIZONTAL C/O	VARIABLE		0.88%		\$122,925	\$12,293
TARGET VERTICAL C/O	VARIABLE		1.90%		\$266,232	\$26,623
TOTAL ALLOCATIONS - FY06 \$ OUTLAYS		100.00%	100.00%	\$11,356,189	\$13,992,000	\$2,534,819

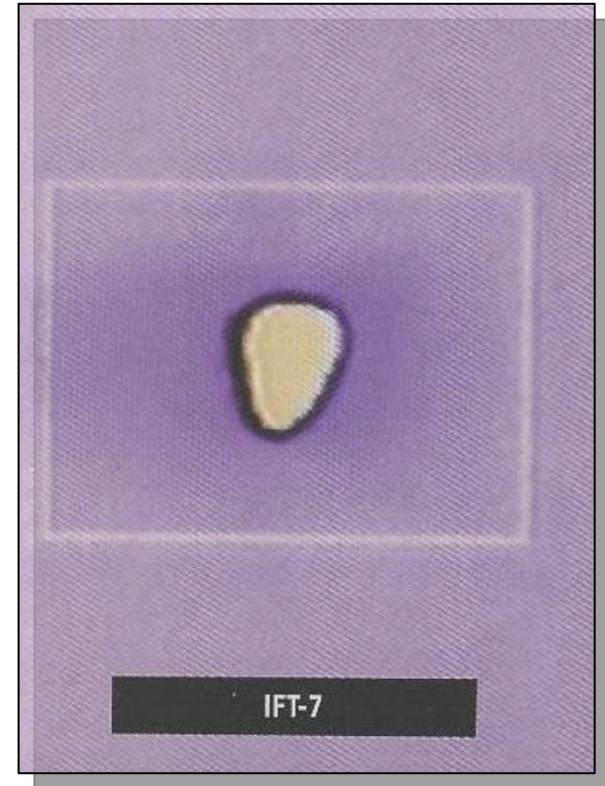
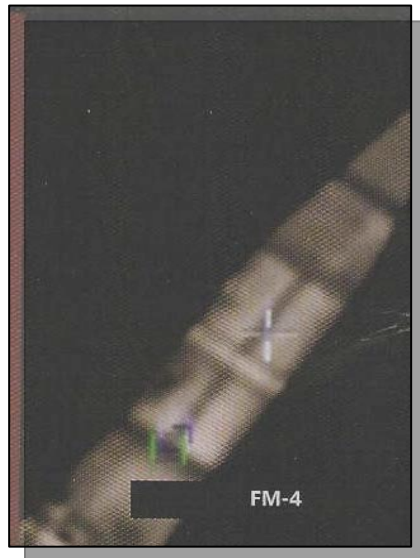
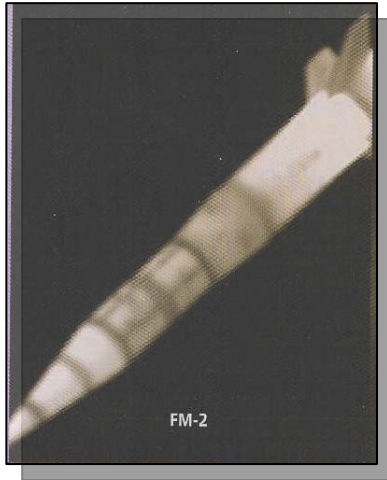
The NASA method results in a Launch Ops Cost Model that forecasts fixed and variable launch costs for vehicles of a certain size. The total fixed and variable costs are then allocated downward to lower WBS levels for budgeting and control purposes.



Missile Cost Modeling – Process Steps

- ***For analogy or bottoms-up, assemble cost data, assemble SMEs.***
- ***For parametric approach, obtain mass properties, SW SLOC, and other drivers for the missiles under study.***
 - ***Define new and inherited content, changes to missile flight software, and test and evaluation program.***
 - ***Revise/tailor CERs and cost models as required.***
- ***Perform and Document Cost excursions :***
 - ***Determine excursions (quantity scenarios, capability excursions, etc.)***
 - ***Document, Present, Defend outputs. Provide recommendations to Program Management.***

Accomplish MDA's Technical and Operational Goals Within Budget



MDA Booster Cost Modeling Conclusions



- ***MDA – Adaptive Organization.. Adaptive Products.***
 - ***Booster products and operational concepts are always changing and are constantly being modified for new applications.***
 - ***Cost models must be revised and updated, as required.***
 - ***MDA policy supports the development of Government/Industry Common Cost Models. (Consensus vs. Confrontation).***
- ***MDA DOBE mission: create and document credible/defendable cost estimates.***