

Rotorcraft Cost Model Enhancements for Future Concepts

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Agenda



- Background
- Data Collection
- TrueRotorcraft Version 1 (Technical Improvements)
- TrueRotorcraft Version 2 (New Rotorcraft Types/Technologies)
- Future Work

The Problem





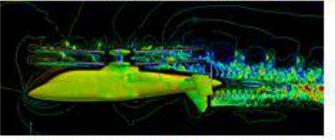


RDECOM

Key Technologies on the Way to JMR & FVL ...



Integrated Aeromechanics and Performance Analysis



HELIOS simulation for Sikorsky coaxial X2TDTM aircraft

Active Flow Control



Description:

Joint DoD High Perf Computing and AMRDEC program for next-gen coupled CFD and CSD to enable accurate multi-fidelity analysis – loads, stability, vibration, acoustics.

S&T Program Objectives:

•97% accuracy for rotor performance
 •95% accuracy for rotor loads
 •95% accuracy for aeroelastic stability
 •10,000x Improvement In computational throughput

Description:

Research and exploratory development of Active Flow Control technology – actuator types, locations,, and control – to delay separation and reduce adverse aerodynamic forces.

S&T Program Objectives:

- 30% fuselage drag/download reduction
- · 20% cruise efficiency (L/De) increase

TECHNOLOGY DRIVEN. WARFIGHTER FOCUSED.

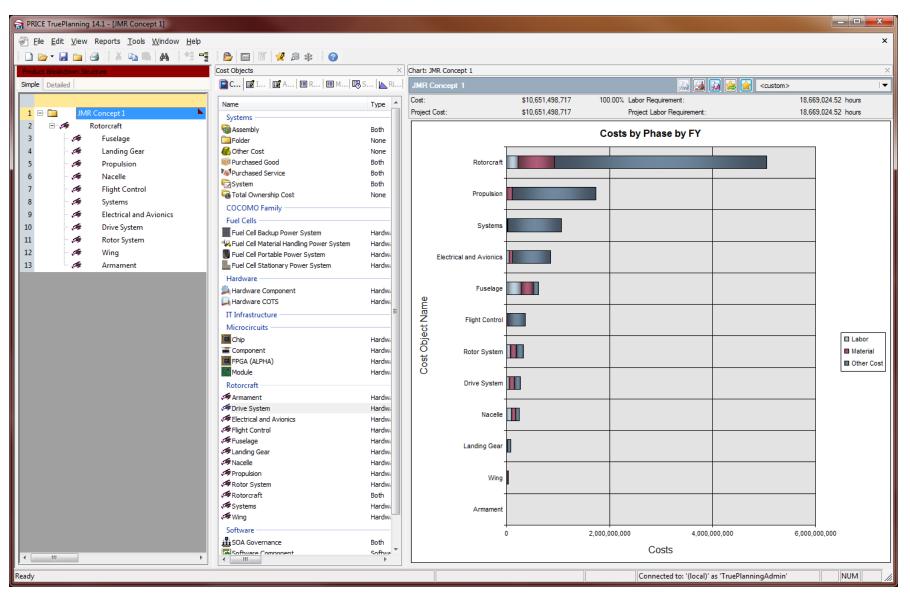
Bell-PC Model



Image: Constraint of the sector of	rl≑ Formulas Data Review View Develo M A R W	Microsoft Excel	- - ×
	Ruler 🛛 Formula Bar 🥄 📑	Hide I Synchronous Scrolling	
Normal Page Page Break Custom Full	Gridlines Headings Zoom 100% Zoom to Selection	New Arrange Freeze Save Switch Macros	
Layout Preview Views Screen Workbook Views	Show Zoom	Vindow All Panes - Unhide Mereset Window Position Workspace Windows	
Material Cost Sum (* fx ="I	Material Cost Summary - Unburdened"&IF(Infla	Factor=1,", 2001\$","")	v
PC_BASED_COST_MODEL_rev_311:3 [Compatibi	B	C_BASED_COST_MODEL_rev_311:1 [Compatibility Mode]	ev_311:2 [Compatibility Mode]
1 A	D	DEVELOPMENT PROGRAM GROUND RULES AND AIRCRAFT CON	Material Cost Summary
2 Project Title:		PROJECT TITLE:	
3			
	oment and Recurring Production Cost	PROGRAM INPUTS: Previous Return to Rate Inputs 3	Tool Raw Material
5	Sment and Recurring Froduction Cost	5 Engineering Design Man-hour Source:	Prediction Adjustment Total Predic \$0 \$0 \$0
6		6 System Weight Source: 6 Wing	
7		7 Number of Prototypes 7 Landing Gear	0 0
8	NEXT	With draw have a more detected in (CTTV) shall be the start (CTTV) 8 Propulsion	
10		and/or fatigue test article (FTA)?	
11	Go To Print Menu	0 Will a wind tunnel test be required?	
12		1 Aircraft Application 12 Electrical/Avionics	0 0
13		2 Alicial Application 1 13 Drive	
15	Go to Data Transfer	3 Category A Certification 15 Armament	
16		4 Will kit development be included in the program cost?	
17		The state of the s	
18	Previous Screen	To Sys Eggin loject Milgin	
20			
21		21 Drive System Bench Tes	st 0 0 0
22	View Results	8 21 Dive System Denci 148	0 0
23		9 AIRCRAFT CONFIGURATION: 23 Data and Manuals	
24 25		0 Aircraft Type 24 Logistics 4 ADULT=+T#=ti=2 25 No GTV, STA, or FTA re	
26		APU installation?	
27 Updated Version: 29 November 2001		2 Landing Gear Type 27 Total Material Cost	\$0 \$0 \$0
28 29		3 Number of engines 28	
30		4 Aircraft Weight Empty 29	Engineering Material
31		30	Prediction Adjustment Total Predic
32		FUSELAGE: 31 Fuselage	\$0 \$0 \$0
33 34		(Includes basic structure, windows, crew doors, passenger doors, cd 32 Wing	0 0 0
35		aft cargo door, floor, tailboom, pylon support, vertical stabilizer, hori 33 Landing Gear 34 Propulsion	
36		6 crew seats, passenger seats, fire extinguisher, soundproofing, and loa 35 Nacelle	
37		7 36 Flight Controls	0 0 0
38 39		8 FUSELAGE CONFIGURATION: 37 Systems 9 Eventore Material 38 Electrical/Avionics	
40		39 Drive	
41		0 Pressunzed Fuselage? 40 Rotor	0 0 0
42 43		1 Crew Door Material 41 Armament	0 0 0
43 H + H START Rate Input Develop	oment Input / Production Input / ProtoProd (Incomposed of the second	Cost Summary Prototy I 4
Ready 🔚			■□□ 100%

TrueRotorcraft

Build PBS from Catalogs of Models to create estimates





Data Collection - Purpose



- Collect and analyze data from in-service rotorcraft
- Create a process for data collection that is:
 - Well defined (to allow apples-to-apples comparison)
 - Repeatable
 - User friendly
- Create a database of Rotorcraft data that represent definitive data that will support future estimates
- Perform analysis on collected data to....
 - Validate existing cost estimating relationships for rotorcraft
 - Identify weaknesses in existing cost estimating relationships and update these based on the analysis

Data Collection Form



- Used the Bell PC model as a starting point
- Extended beyond the Bell PC model to accommodate for new technologies, materials, and anything Army/Industry suggested may drive cost.
- Data collection form has tabs representing the following data that might be available
 - Aircraft and Subsystem
 - Weight breakdown by subsystem (via Bell PC model and Society of Allied Weight Engineers standard RP-8A)
 - Cost and Effort Summary
 - Development Cost Details
 - Prototype Details
 - Production Details
 - Reliability and Maintainability
 - Maintenance Manhours and Parts cost
 - Other O&S Data
 - Detailed Subsystem Data

Data Collection

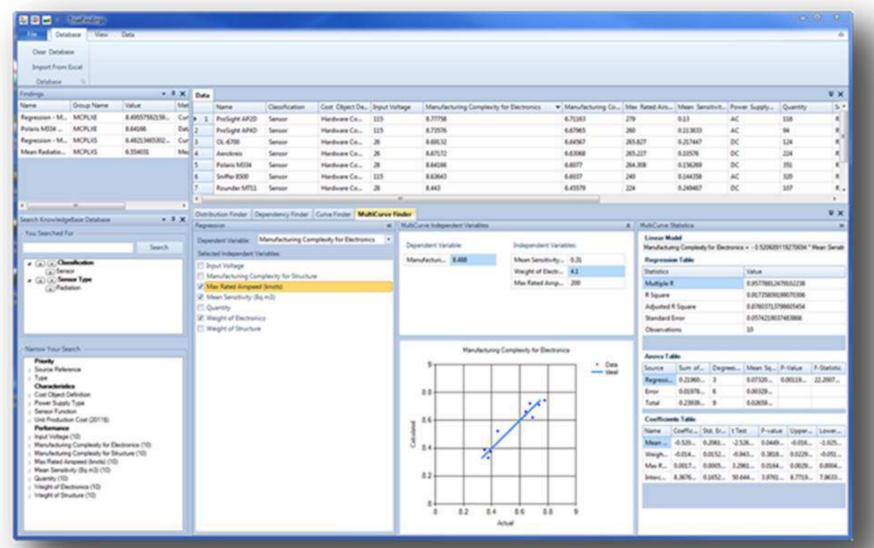


- Data Collection Forms distributed to Army and OEMS
 - Actual data collection has been a slow starter but looks promising going forward
- Plan B being executed gathering CSDR data directly from Army, and matching up as best as possible to data collection forms.

Rotorcraft Cost Database



use TrueFindings for statistics to support model parameters, CERs and Results



TrueRotorcraft v1.0 Improvements



- User friendly, visual, drag-n-drop, plug-n-play environment
- Rapid tradeoff & input sensitivity analysis
- Years of Planned Production capability, O&S Deployment, and basic Schedule Estimating Relationships
- Enhanced reporting features
- New capabilities for economic considerations
 - Inflation data built-in, or can be supplied by the estimator
 - View costs as-spent, or in any specific FY
 - Net Present Value
 - Costs vs. Budget
- Complete Review of Algorithms
 - 10+ cost-changing bugs found in original model, fixes ready for v2.0

- Automated Testing

- Over 1000 automated tests and counting
- Enables updating /enhancing of models with greatly reduced danger of introducing bugs in existing algorithms.
- Improved Help System

TrueRotorcraft v2.0 Improvements



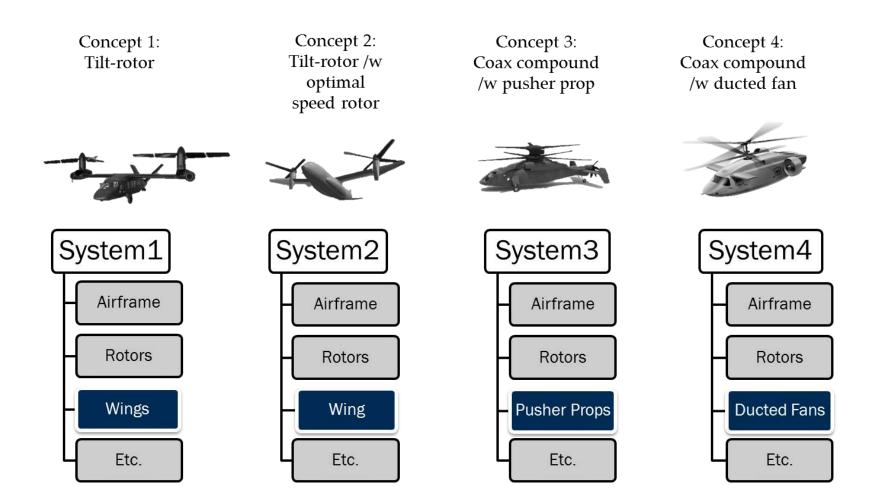
- Introduce Component Build-up Capability
- Improve Technology Factors (Georgia Tech PhD's provide research and guidance)
- O&S Model Move to MTBF-Driven Methodology (Georgia Tech)
- Map MIL-STD-881C and CAPE Cost Element Structure

Recommendations for Model Improvement

Recommended Changes	Schedule Risk	Benefits
1. Component build-up capability		*****
2. Complexity factor calculator	***	****
3. Technology profile	***	****
4. RAM Allocation	****	*****
5. Update O&S WBS	******	****
6. Software module	*******	******

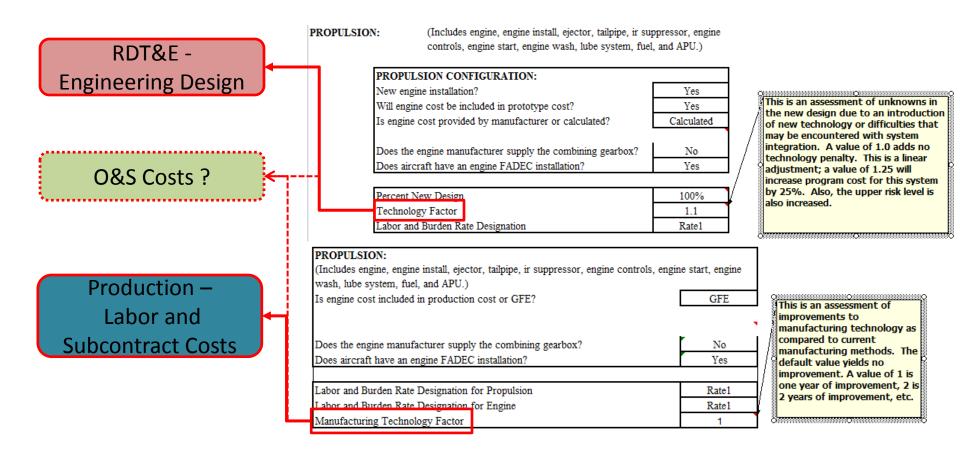
Component Build Up





Current Use of Technology Factor

- Technology Factors and Manufacturing Technology Factors are used to account for increased RDT&E and Production costs due to advanced technology
- However, these factors cannot be used to influence O&S costs



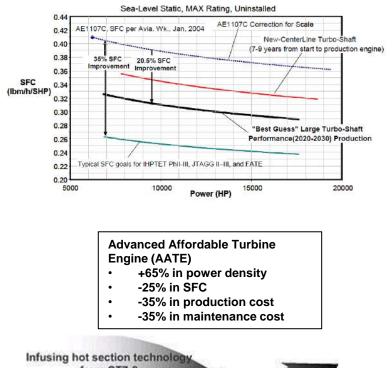
Technology Factor



Trends of Advanced Technology

- Provide realistic boundaries to quantify technology factors guided by actual data
- Include a range of possibility -- optimistic/most likely/pessimistic
- Impacts throughout life cycle can be defined
- Technology database
 - User-define technology impacts
 - Include specific S&T programs or any product development efforts
 - Establish risk level based on technology trends, schedule, TRL, and etc.

UH-60M baseline: GE T700-701D									
System Attribute	A	ATE Goals		Actual	Note				
R&D Cost		117.72		0	\$M FY2013				
SFC		0.3465		0.462	At max. power				
T/W		6.8376		4.144	At max. power				
Engines Price	\$	1,056,965	\$	1,626,100	FY2013 DoD Budget				
O&S (engine only)	\$	780.19	\$	1,200	FY2013/FH Estimated using Harris '12				
TRL		6		9	Assumed				





Database of Technology Candidates



DATA	SOURCE CHIEGO	CHOD AT LOWAN	SSES MARACT	BAGELEMENTS BASELINE	MRCRAFT BASS	LINE PERFORMAN	NED PERFORMENT	ABLE THE	EFRANCE FRANK	E FACTOR A RD	Sat R	5 ¹ 0 ⁸	5 84
BOEING STUDY	ENGINE FUEL FLOW		SEC	AE1107C ENGINE ON V-22	0.41 Ibm/h/h p		2020-2030		9	-	-	-	2001
FATE	ENGINE POWER DENSITY	INCREASE POWER/WEIGHT RATIO	REDUCE EMPTY WEIGHT	GE T700-701D ON BLACKHAWK	100%	35.00%	2015		6	-	-	-	2001
AATE	ENGINE	ΔΔΤΕ	SFC T/W	GE T700-701D ON BLACKHAWK	0.3465 6.8376	-35% 65%	2013		6	117.7 2\$M	-35%	-35%	2013
BOEING STUDY	STRUCTURAL WEIGHT	ADVANCED MATERIALS, AIRFRAME DESIGN AND OPTIMIZATION, AND MANUFACTURING TECHNIQUES FOR LIGHTER STRUCTURE	AIRFRAME WEIGHT	V-22 AIRFRAME	100%	-17.80%	2023		9	-	-	-	2001
BOEING STUDY	DRIVE SYSTEM	PROCESS	DRIVE SYSTEM WEIGHT	SINGLE SPEED XMSN	100%	-30%	2023		9	35%	15%	-30%	2009
BOEING STUDY	DRIVE SYSTEM	DRIVE CONFIG, ADV COMPONENT, MAT. AND PROCESS, INFRASTRUCTURE AND SUPPORT SYSTEM	DRIVE SYSTEM WEIGHT	VARIABLE SPEED XMSN	100%	-22.00%	2023		9	45%	25%	5%	2009
BOEING STUDY	ROTOR TECHNOLOGY		ROTOR BLADE WEIGHT	V-22 ROTOR BLADE	100%	-15%	2023		9	-	-	-	2001

Technology Factor Implementation



31	AATE Program				
32	Include AATE Costs/Benefits?	Yes 🔽		0	
33	AATE R&D Cost	117,720,000	s 💷	0	1
34	Fuel Consumption Factor	0.750		0	1
35	Engine Price Factor	0.650		N	1
36	AATE Maintenance Cost Factor (Parts and Overhaul)	0.650		0	

Cross Project Reports					B	
Name / UH-60M AATE Modification	Owner TruePlanningAdmin TruePlanningAdmin	Re	sults Chart			
				Total	UH-60M AATE Modification	UH-60M Baseline
			1 Development	1,339,321,580	748,456,557	590,865,024
			2 Production	9,779,515,038	4,710,338,604	5,069,176,434
			3 Operation & Support	15,714,767,025	7,710,741,272	8,004,025,753
			4 Total	26,833,603,643	13,169,536,432	13,664,067,211

Enhance Methodology to be MTBF-driven



RAM Input Option 1 : System Level RAM Allocation Option 2 : Sub-system level input

			-
0	noration	al Innuta	
0	peration	al Inputs	

Peacetime	Month	Year	
Flying Hours/AC (Operational)	18	216	
Flying Hours/AC (TDA)	30	360	
ALDT (Peacetime hours)	22		
Wartime	Month	Year	
¥artime Flying Hours/AC (Operational)	Month 183	Year 2196	

Reliability Input				
System	Component Name	МТВЕМА		
		Predicted		
	WING GROUP	-		
	ROTOR GROUP	15.226		
	EMPENNAGE GROUP			
	FUSELAGE GROUP	18.806		
	ALIGHTING GROUP	60.463		
	ENGINE SECTION OR NACELLE GROUP	52.756		
	AIR INDUCTION GROUP	15176.200		
	PROPULSION GROUP	44.322		
	AUXILIARY POWER GROUP	1785.435		
icle	FLIGHT CONTROLS GROUP	68.533		
Air Vehicle	INSTRUMENT GROUP	240.072		
Air	HYDRAULIC GROUP	97.596		
	PNEUMATIC GROUP			
	ELECTRICAL GROUP	66.687		
	AVIONICS GROUP	84.277		
	ARMAMENT GROUP			
	FURNISHINGS & EQUIPMENT GROUP	63.462		
	ENVIRONMENTAL CONTROL GROUP	2385.693		
	PHOTOGRAPHIC GROUP			
	LOAD & HANDLING GROUP	155.285		
	ANTI - ICING GROUP	194.567		
	MTBEMA Adjustment Factor	1.000		
	Total Aircraft MTBEMA	3.829		

Maintainability Input				
vstem	Component Name	MTTRe		
<i>Jetem</i>		Predicted		
	WING GROUP			
	ROTOR GROUP	1.526		
	EMPENNAGE GROUP			
	FUSELAGE GROUP	1.705		
	ALIGHTING GROUP	1.558		
	ENGINE SECTION OR NACELLE GROUP	2.820		
	AIR INDUCTION GROUP	0.683		
	PROPULSION GROUP	2.082		
	AUXILIARY POWER GROUP	2.540		
Air Vehicle	FLIGHT CONTROLS GROUP	2.275		
Veh	INSTRUMENT GROUP	0.549		
Air	HYDRAULIC GROUP	1.650		
	PNEUMATIC GROUP			
	ELECTRICAL GROUP	1.228		
	AVIONICS GROUP	0.827		
	ARMAMENT GROUP			
	FURNISHINGS & EQUIPMENT GROUP	0.953		
	ENVIRONMENTAL CONTROL GROUP	0.689		
	PHOTOGRAPHIC GROUP			
	LOAD & HANDLING GROUP	1.233		
	ANTI - ICING GROUP	1.626		
	MTTRe Adjustment Factor	1.000		
	Total Aircraft MTTRe	1.650		

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Mapping to MIL-STD-881C and CAPE O&S CES

One-to-one mapping between existing Bell PC Model and CAPE cost structure

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Questions?



